

### TRANSMITTAL

<b>TO:</b>	Fermi Research Alliance LLC.	<b>DATE:</b>	9-3-19
		<b>Transmittal No.</b>	8018-195-T-005
		<b>RE:</b>	100% Complete Cage Drawings
<b>ATTN:</b>	David L. Crabb		

**WE ARE SENDING YOU:**    ☒ Attached the following items:    ☐ Under separate cover via: \_\_\_\_\_

☒ Design Drawings    ☐ Prints    ☐ Plans    ☐ Samples  
☒ Specifications    ☐ Copy of Letter    ☐ Change Order    ☐ Electronic Drawing

COPIES	DATE	DRAWING NO.	SHEET NO.	REV.	DESCRIPTION
	9-3-19	8018195-320-01	1-4	P-6	Preliminary Approval cage design drawings
	9-3-19	8018195-320-02	1-3	P0, P4, P5	Preliminary Approval work deck design drawings
	9-3-19		1-8		Lifting Lug Design Calculations
	9-3-19		1-3		Cage Floor Main Beam Calculations
	9-3-19		1-9		Underslung lug design #1 Calculations
	9-3-19		1-9		Underslung lug design #2 Calculations
	9-3-19		1-9	REV A	Structural Design Calculations

**THESE ARE TRANSMITTED as checked below:**

- |   |   |   |
|---|---|---|
| <input checked="" type="checkbox"/> For Approval  | <input type="checkbox"/> Approved as submitted    | <input type="checkbox"/> Resubmit _____ copies for approval   |
| <input type="checkbox"/> For your Use             | <input type="checkbox"/> Approved as noted        | <input type="checkbox"/> Submit _____ copies for distribution |
| <input type="checkbox"/> As requested             | <input type="checkbox"/> Returned for corrections | <input type="checkbox"/> Return _____ corrected prints        |
| <input type="checkbox"/> For review and comment   | <input type="checkbox"/> Other:                   |   |
| <input type="checkbox"/> FOR BIDS DUE _____, 20__ |   | <input type="checkbox"/> PRINTS RETURNED AFTER LOAN TO FKCI   |

**REMARKS:**

<b>COPY TO:</b>	Syd Devries
<b>SIGNED:</b>	Kevin Dartt
<b>DATE:</b>	9-3-19

8018195-320-23  
WEAR SHOE SHIM  
8018195-320-22  
WEAR SHOE

8018195-320-21  
WEAR SHOE

8018195-320-08  
DIAGONAL BRACE

8018195-320-16  
WALL PANEL

8018195-320-17  
WALL PANEL

8018195-320-29  
ROOF SUPPORT

8018195-320-10  
TOP TRANSOM

8018195-320-11  
CAGE DOGGING

8018195-320-13  
ROOF DECK

8018195-320-12  
ROOF DECK

8018195-320-27  
ROLL-UP DOOR ASS'Y

8018195-320-07  
BAIL STRINGER

FIRE EXTINGUISHER

8018195-320-TG  
EQUIPMENT TAG

8018195-320-41  
WINDOW DOOR

FLOOD LIGHT

8018195-320-30  
CHAIRING MECHANISM

8018195-320-03  
BOTTOM DECK

8018195-320-18  
SLINGING LUG

8018195-320-19  
SLINGING LUG

8018195-320-24  
SLINGING LUG

8018195-320-05  
FLOOR RAIL

8018195-320-01  
LINK BAR

HW04-0028  
LINK PIN

8018195-320-12  
ROOF DECK

WARNING PLACARD

8018195-320-20  
LIFTING LUG

8018195-320-32  
COUNTER WT. GUARD

8018195-320-06  
CORNER STRINGER

8018195-320-35  
ELECT. ENCL. BOX

8018195-320-09  
DIAGONAL BRACE

8018195-320-15  
WALL PANEL

GD07-0005  
12.5" GUIDE WHEEL ASSY

8018195-320-14  
WALL PANEL - R.H. FRONT

8018195-320-76  
LOAD SECURING BAR

FOR APPROVAL

DATE: \_\_\_\_\_

APP. BY: \_\_\_\_\_

TOTAL WEIGHT (LB)

8457

NOTE:

1. TOTAL WEIGHT DOES NOT INCLUDE INSPECTION CANOPY/HANDRAIL, MAINTENANCE PLATFORM/CANOPY/HANDRAIL.

2. ALL FASTENERS TO BE STAINLESS STEEL

41	4		FLOOD LIGHT	FLOOD LIGHT			4
40	1		FIRE EXTINGUISHER	20 LB. CLASS ABC			31
39	2	1142A330	SEAL	3/8" X 5/8" RUBBER SEAL			0
38	8	94748A426	PIN	QUICK RELEASE PIN			2
37	1	HW04-0028	LINK PIN	SEE ASSEMBLY DRAWING			6
36	4	GD07-0005	12.5" GUIDE WHEEL ASSY	SEE ASSEMBLY DRAWING			1148
35	1	CS6721	SOCKET	M26 SOCKET W/ PIN			36
34	1	8018195-340-01	LINK BAR	SEE DETAIL DRAWING			92
33	1	8018195-320-TG	EQUIPMENT TAG	SEE DETAIL DRAWING			0
32	4	8018195-320-76	LOAD SECURING BAR	SEE ASSEMBLY DRAWING			38
31	2	8018195-320-41	WINDOW DOOR	SEE ASSEMBLY DRAWING			21
30	1	8018195-320-40	ESCAPE DOOR	SEE ASSEMBLY DRAWING			39
29	2	8018195-320-36	INSPECTION CANOPY	SEE ASSEMBLY DRAWING			728
28	2	8018195-320-35	ELECT. ENCL. BOX	SEE ASSEMBLY DRAWING			32
27	4	8018195-320-32	COUNTER WT. GUARD	SEE ASSEMBLY DRAWING			143
26	1	8018195-320-30	CHAIRING MECHANISM	SEE ASSEMBLY DRAWING			574
25	2	8018195-320-29	ROOF SUPPORT	SEE ASSEMBLY DRAWING			48
24	1	8018195-320-27	ROLL-UP DOOR ASS'Y	SEE ASSEMBLY DRAWING			503
23	1	8018195-320-26	FLOOR - KICK PL.	SEE ASSEMBLY DRAWING			9
22	1	8018195-320-24	SLINGING LUG	SEE ASSEMBLY DRAWING			24
21	4	8018195-320-23	WEAR SHOE SHIM	SEE ASSEMBLY DRAWING			89
20	4	8018195-320-22	WEAR SHOE	SEE ASSEMBLY DRAWING			130
19	8	8018195-320-21	WEAR SHOE	SEE ASSEMBLY DRAWING			81
18	4	8018195-320-20	LIFTING LUG	SEE ASSEMBLY DRAWING			21
17	6	8018195-320-19	SLINGING LUG	SEE ASSEMBLY DRAWING			143
16	1	8018195-320-18	SLINGING LUG	SEE ASSEMBLY DRAWING			32
15	1	8018195-320-17	WALL PANEL	SEE ASSEMBLY DRAWING			158
14	1	8018195-320-16	WALL PANEL	SEE ASSEMBLY DRAWING			153
13	1	8018195-320-15	WALL PANEL	SEE ASSEMBLY DRAWING			155
12	1	8018195-320-14	WALL PANEL - R.H. FRONT	SEE ASSEMBLY DRAWING			151
11	1	8018195-320-13	ROOF DECK	SEE ASSEMBLY DRAWING			120
10	2	8018195-320-12	ROOF DECK	SEE ASSEMBLY DRAWING			353
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8	1	8018195-320-10	TOP TRANSOM	SEE ASSEMBLY DRAWING			737
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6	2	8018195-320-08	DIAGONAL BRACE	SEE ASSEMBLY DRAWING			144
5	2	8018195-320-07	BAIL STRINGER	SEE ASSEMBLY DRAWING			318
4	4	8018195-320-06	CORNER STRINGER	SEE ASSEMBLY DRAWING			150
3	2	8018195-320-05	FLOOR RAIL	SEE ASSEMBLY DRAWING			453
2	1	8018195-320-04	FLOOR TIMBERS	SEE ASSEMBLY DRAWING			475
1	1	8018195-320-03	BOTTOM DECK	SEE ASSEMBLY DRAWING			1131
ITEM	QTY	PART NO.	NOMENCLATURE	DESCRIPTION	MATERIAL	COMMENTS	WT. (LB)

8018195-320-01  
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HW04-0028  
LINK PIN

8018195-320-12  
ROOF DECK

WARNING PLACARD

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8018195-320-32  
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TOTAL WEIGHT (LB)

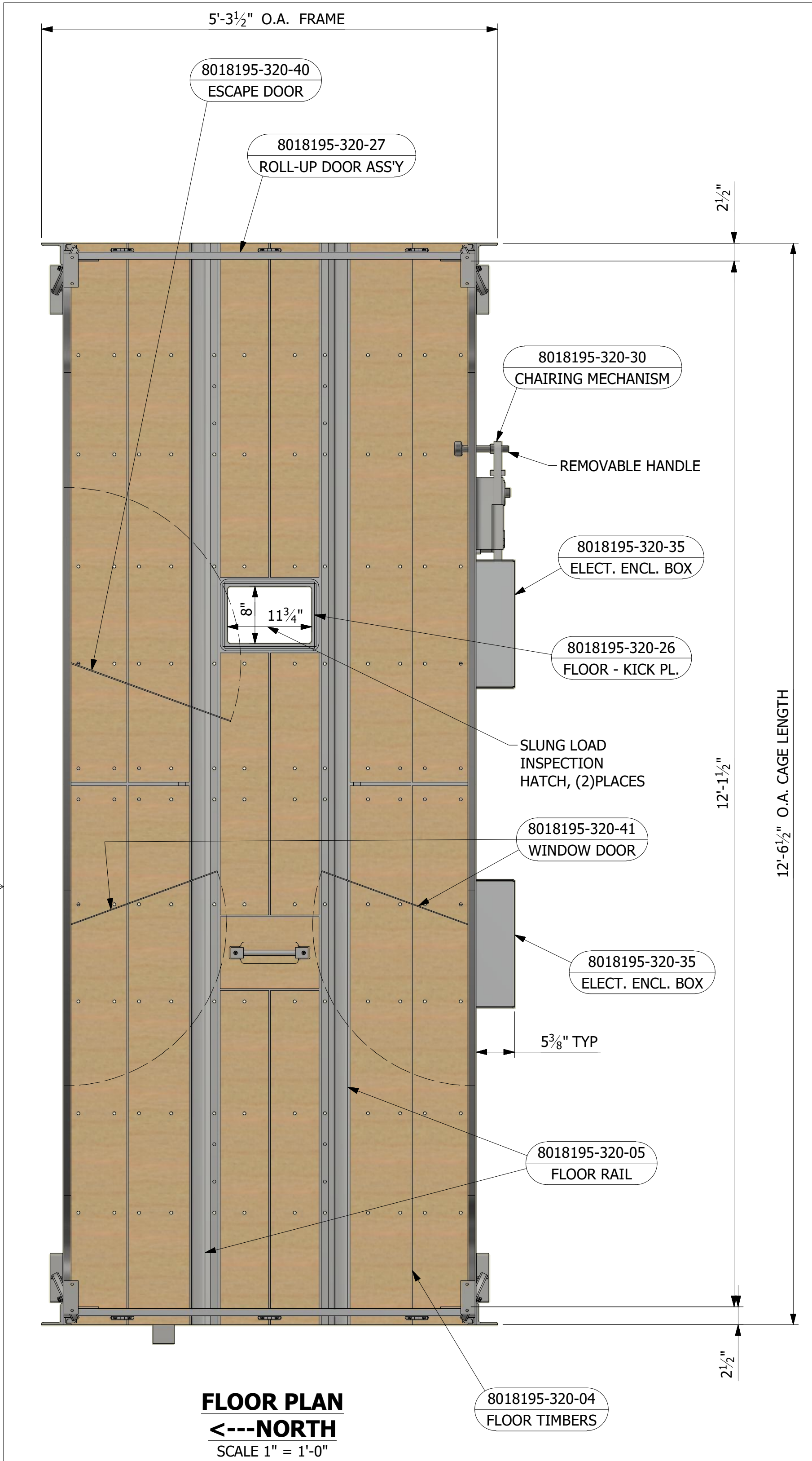
8457

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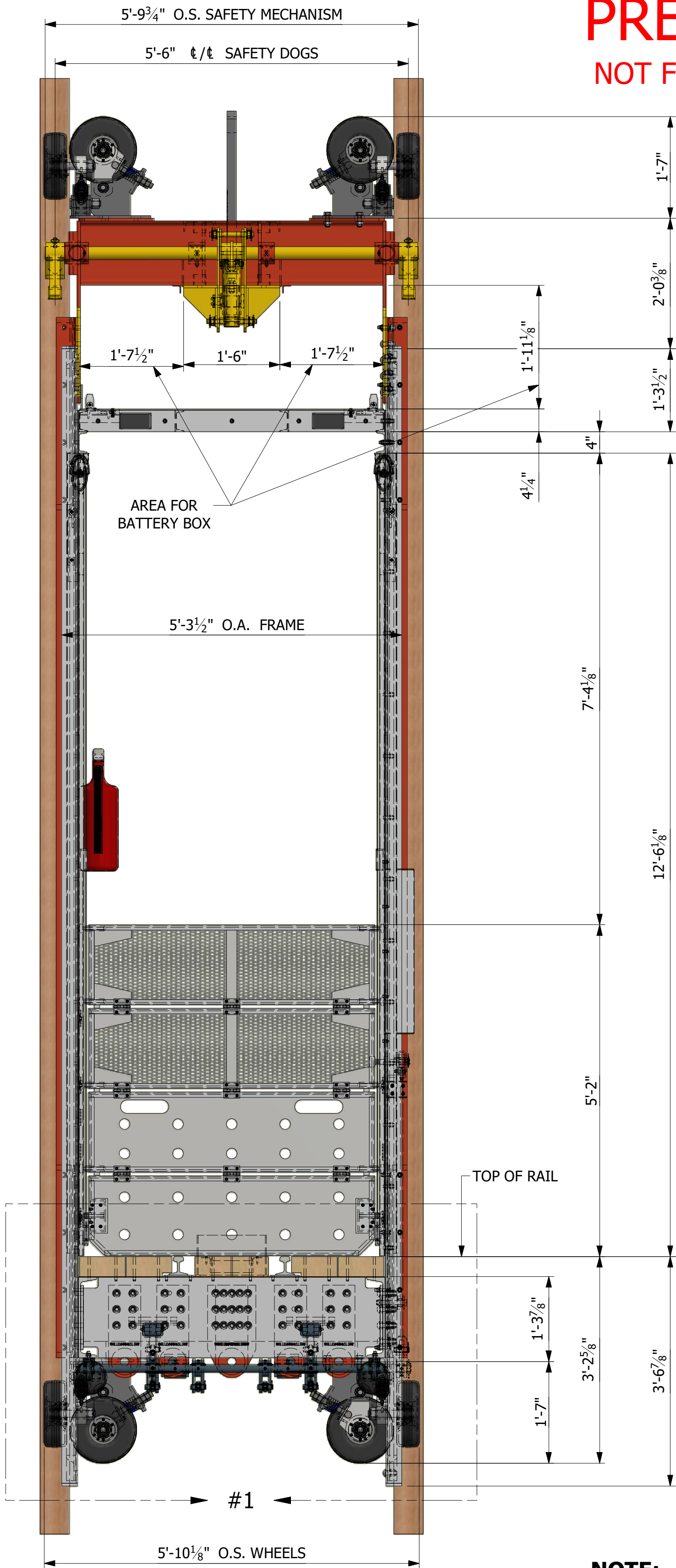
PROPRIETARY NOTICE: THIS DOCUMENT OR DRAWING AND ALL INFORMATION THEREON IS THE PROPERTY OF FRONTIER-KEMPER CONSTRUCTORS OF EVANSVILLE, IN. NEITHER RECEIPT NOR POSSESSION THEREOF IMPLIES OR TRANSFERS ANY RIGHT IN, OR LICENSE TO USE THE DOCUMENT, SUBJECT MATTER, DESIGN, OR INFORMATION SHOWN THEREON, OR ANY RIGHT TO REPRODUCE THIS DOCUMENT, OR ANY PART THEREOF. NEITHER THIS DOCUMENT NOR ANY INFORMATION CONTAINED THEREIN MAY BE COPIED OR REPRODUCED, IN WHOLE OR IN PART, OR OTHERWISE USED WITHOUT THE EXPRESSED WRITTEN PERMISSION FROM FRONTIER-KEMPER CONSTRUCTORS, INC.

UNLESS OTHERWISE SPECIFIED, ALL DIMENSIONS ARE IN INCHES

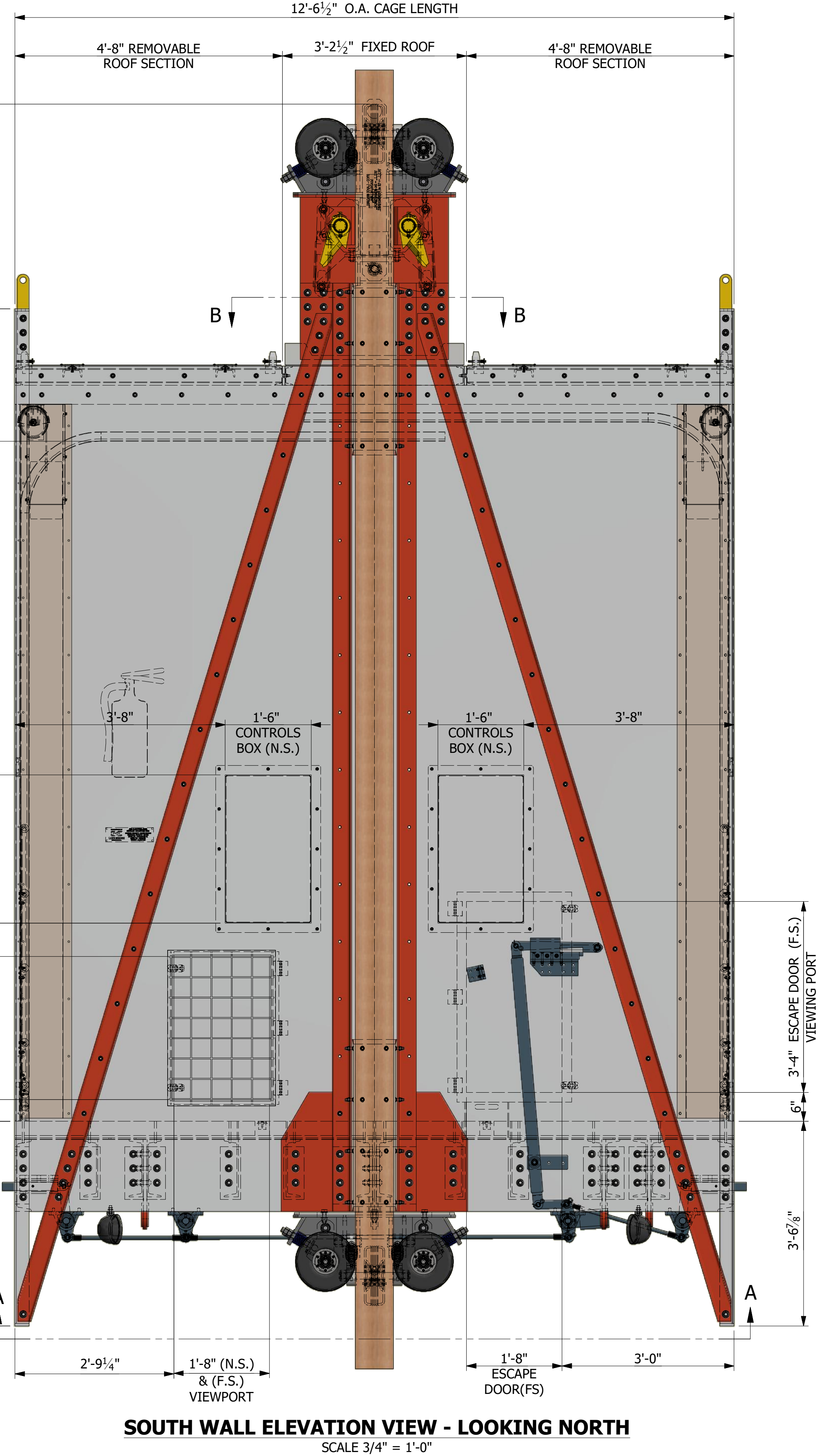
**TOLERANCES:**  
FRACTIONAL: ± 1/16"  
ANGULAR MACHINE = ± 0.5°  
ANGULAR BEND = ± 1°  
X.XX = ± 0.01  
X.XXX = ± 0.005  
X.XXXX = ± 0.0005

INTERPRET DRAWING IN ACCORDANCE WITH STANDARDS PRESCRIBED IN ASME Y14.100M

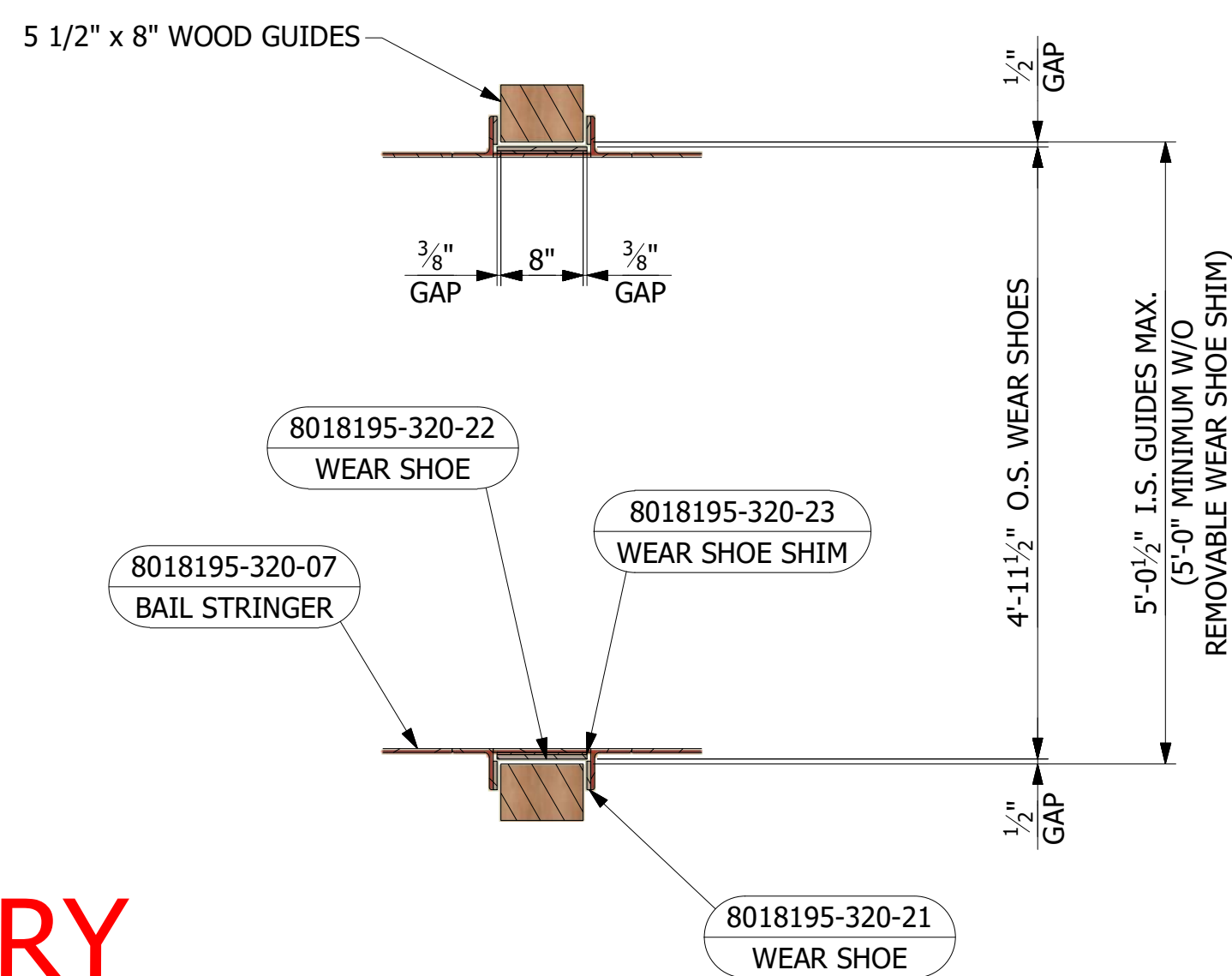
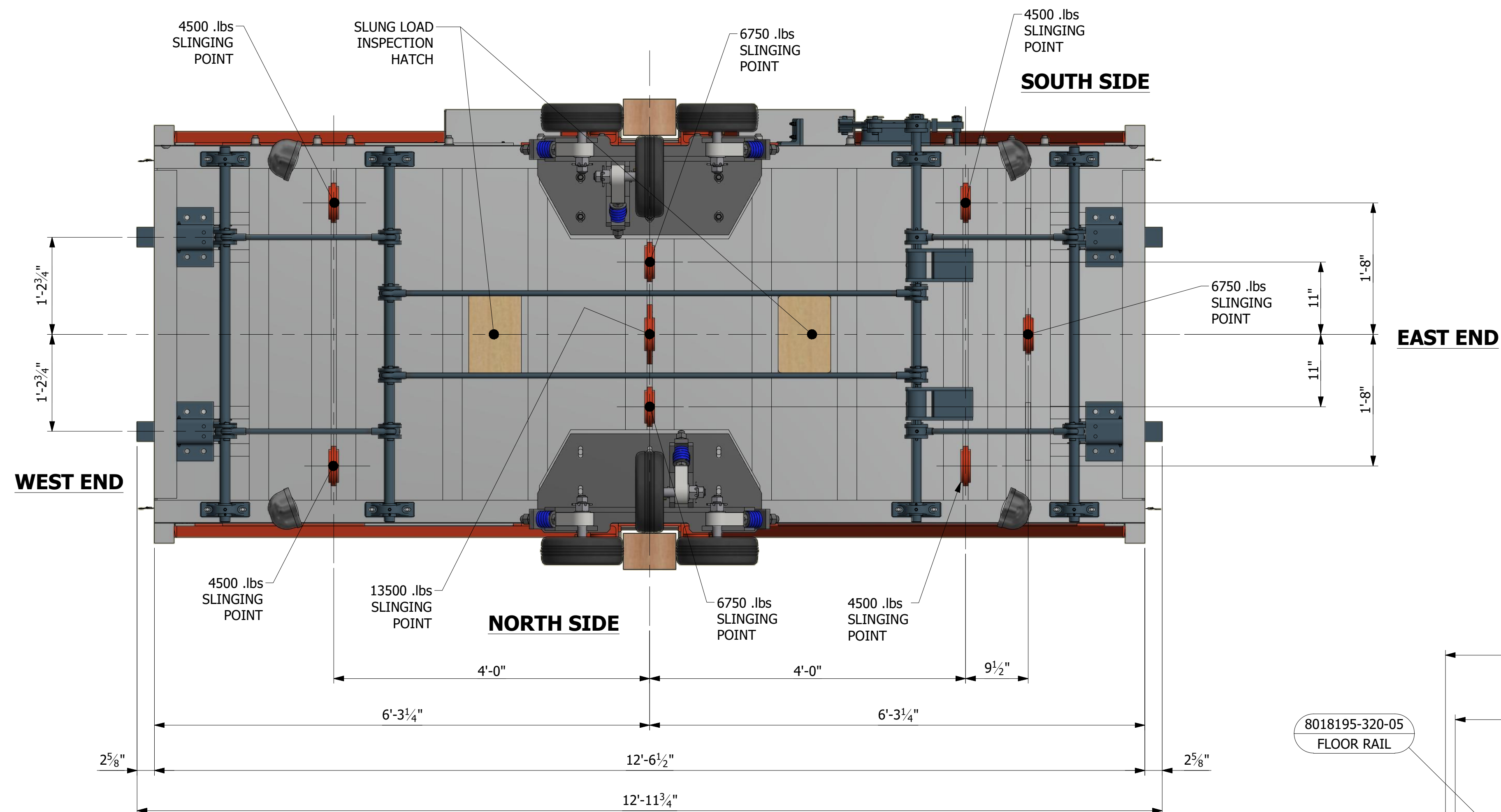
REV.	DATE	BY	DESCRIPTION	APP.	REV.	DATE	BY	DESCRIPTION	APP.	DWG DATE:
P1	10/15/18	JAD	GENERAL UPDATE FOR PROGRESS	CLL	P4	11/13/18	JAD	ISSUED FOR 75% DESIGN REVIEW	CLL	9/4/2018
P2	10/25/18	JAD	UPDATES FOR DESIGN CHANGE REQUEST	CLL	P5	6/5/19	JAD	ISSUED FOR 90% DESIGN REVIEW	CLL	DESIGN BY: FKC - LS
P3	11/5/18	JAD	UPDATES FOR DESIGN CHANGE REQUEST	CLL	P6	8/30/2019	MAK	ISSUED FOR 100% APPROVAL	CLL	DRAWN BY: JAD
									CLL	CHECKED BY: CLL



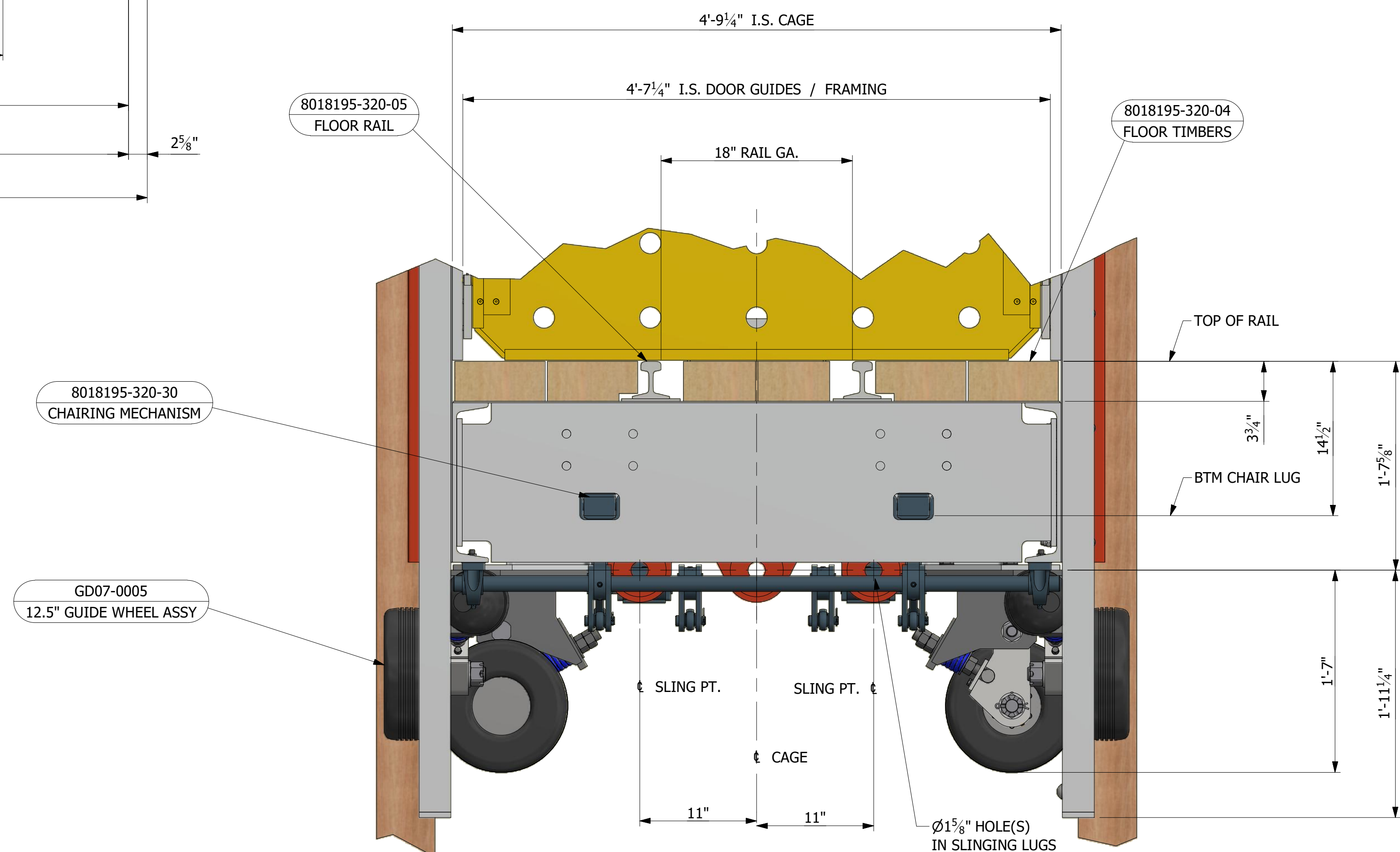
**NOTE:**  
1.) SEE ADDITIONAL SECTIONS & DETAILS ON SHEET #3.  
2.) SEE INSPECTION DECK ARRANGEMENT ON SHEET #4.



CLIENT	FKC - LAKESHORE EVANSVILLE, IN MAN / MATERIAL CAGE	TITLE	ROSS MAN / MATERIAL CAGE GENERAL ARRANGEMENT	FILE: 8018195-320-01.idw
DRAWING STATUS: PRELIMINARY	ORIGINAL SIZE: 22 x 34	DRAWING NO.	8018195-320-01	REV. P6
		DO NOT SCALE DRAWING	SHEET: 2 OF 4	



**PRELIMINARY**  
**NOT FOR CONSTRUCTION**



**VIEW A-A**  
**SLINGING POINTS**  
SCALE 1" = 1'-0"

**NOTE:** ALL SLINGING POINT LUGS  
STAMPED WITH LOAD RATING

FOR APPROVAL

DATE: \_\_\_\_\_

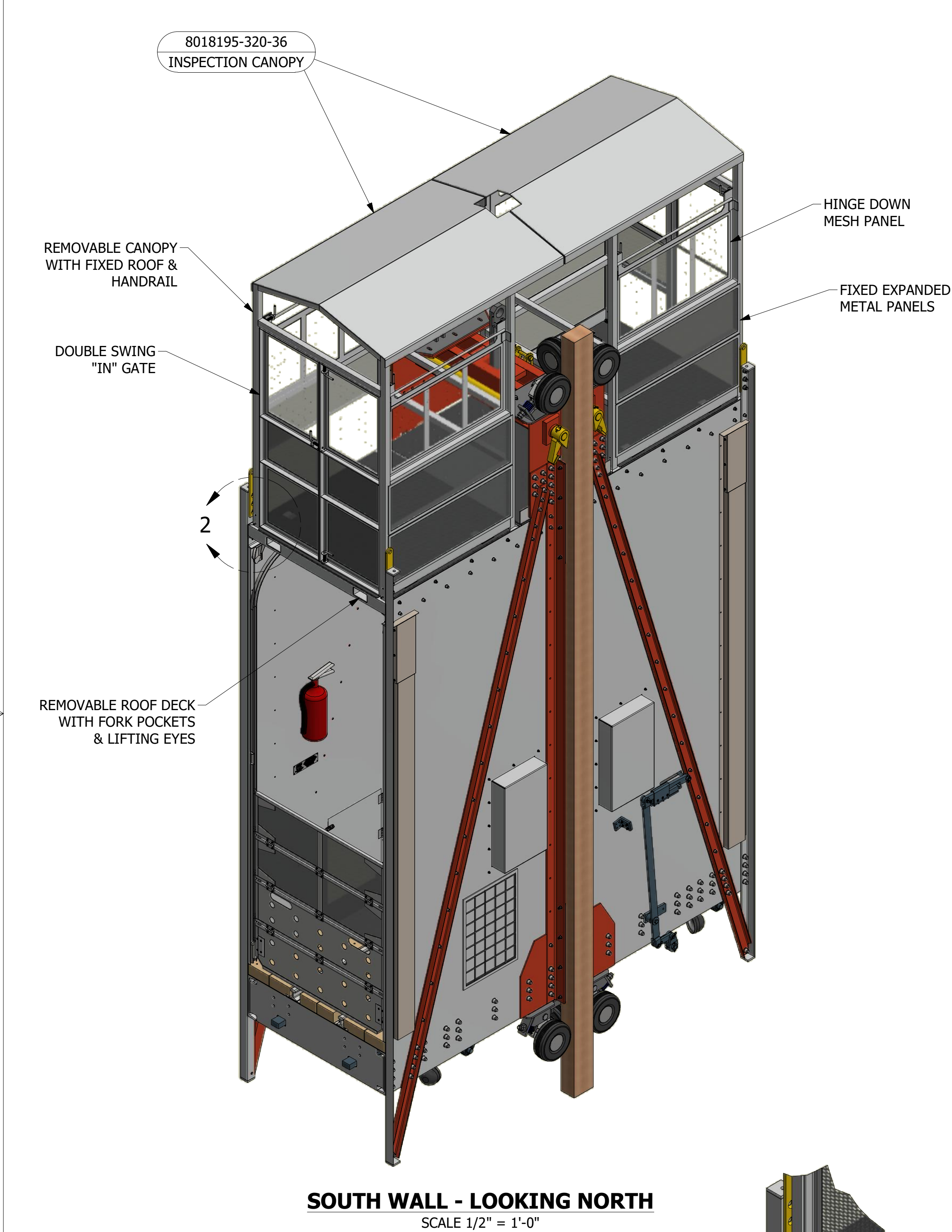
APP. BY: \_\_\_\_\_

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P1	10/15/18	JAD	GENERAL UPDATE FOR PROGRESS	CLL	P4	11/13/18	JAD	ISSUED FOR 75% DESIGN REVIEW	CLL	9/4/2018
P2	10/25/18	JAD	UPDATES FOR DESIGN CHANGE REQUEST	CLL	P5	6/5/19	JAD	ISSUED FOR 90% DESIGN REVIEW	CLL	DESIGN BY: FKC - LS
P3	11/5/18	JAD	UPDATES FOR DESIGN CHANGE REQUEST	CLL	P6	8/30/2019	MAK	ISSUED FOR 100% APPROVAL	CLL	DRAWN BY: JAD
									CHECKED BY: CLL	



CLIENT	FKC - LAKESHORE EVANSVILLE, IN MAN / MATERIAL CAGE		TITLE  ROSS MAN / MATERIAL CAGE SECTIONS AND DETAILS	FILE: 8018195-320-01.idw		
				DRAWING NO. 8018195-320-01		REV. P6
	DRAWING STATUS: PRELIMINARY	ORIGINAL SIZE: 22 x 34		DO NOT SCALE DRAWING		SHEET: 3 OF 4



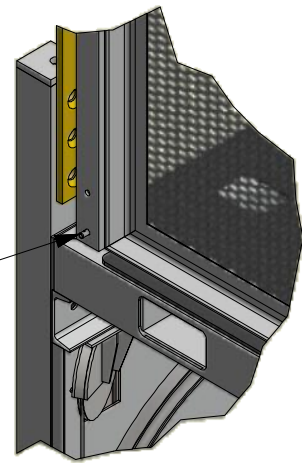
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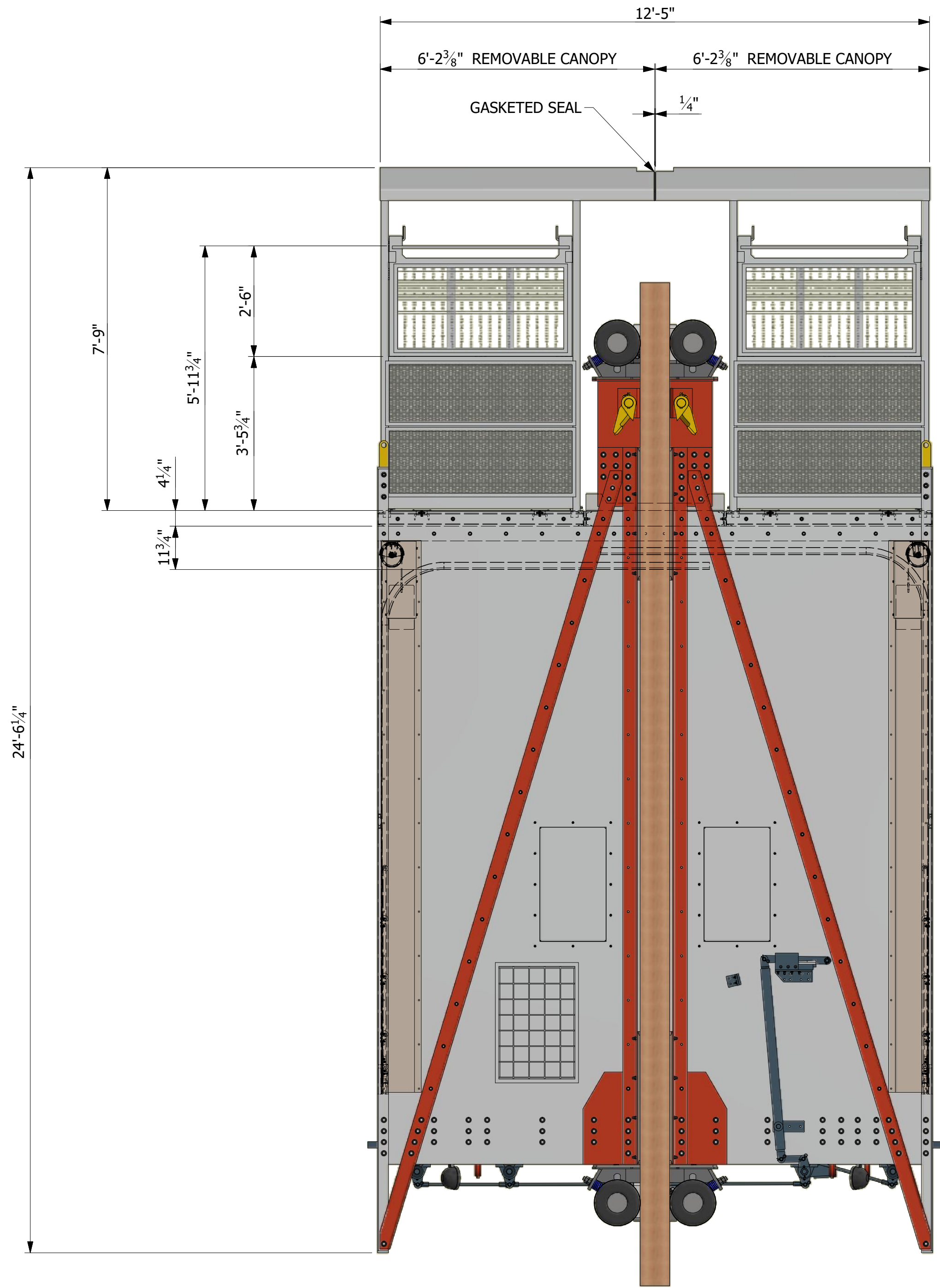
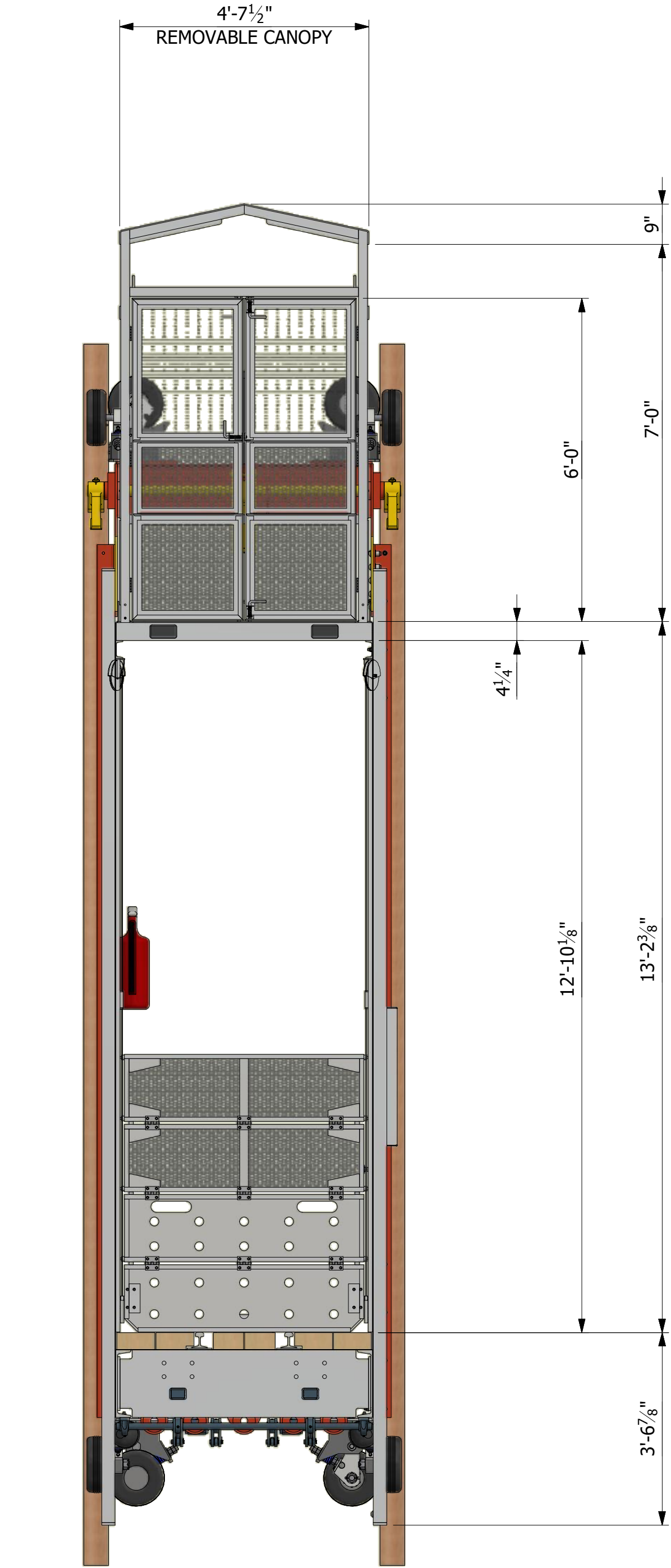
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INTERPRET DRAWING IN ACCORDANCE WITH STANDARDS PRESCRIBED IN ASME Y14.100M

94748A426  
PIN



**DETAIL 2**  
SCALE 1" = 1'-0"



**PRELIMINARY**  
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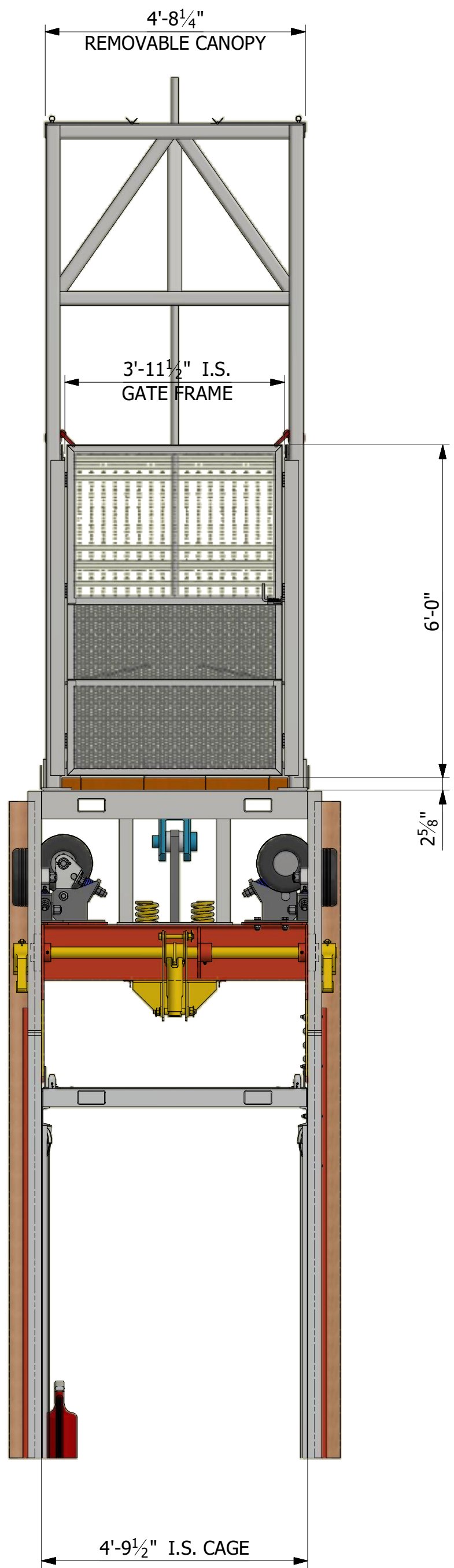
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DRAWING STATUS: PRELIMINARY	ORIGINAL SIZE: 22 x 34	DRAWING NO.	8018195-320-01	REV. P6
		DO NOT SCALE DRAWING	SHEET: 4 OF 4	



**WEST END VIEW - LOOKING EAST**  
SCALE 1/2" = 1'-0"

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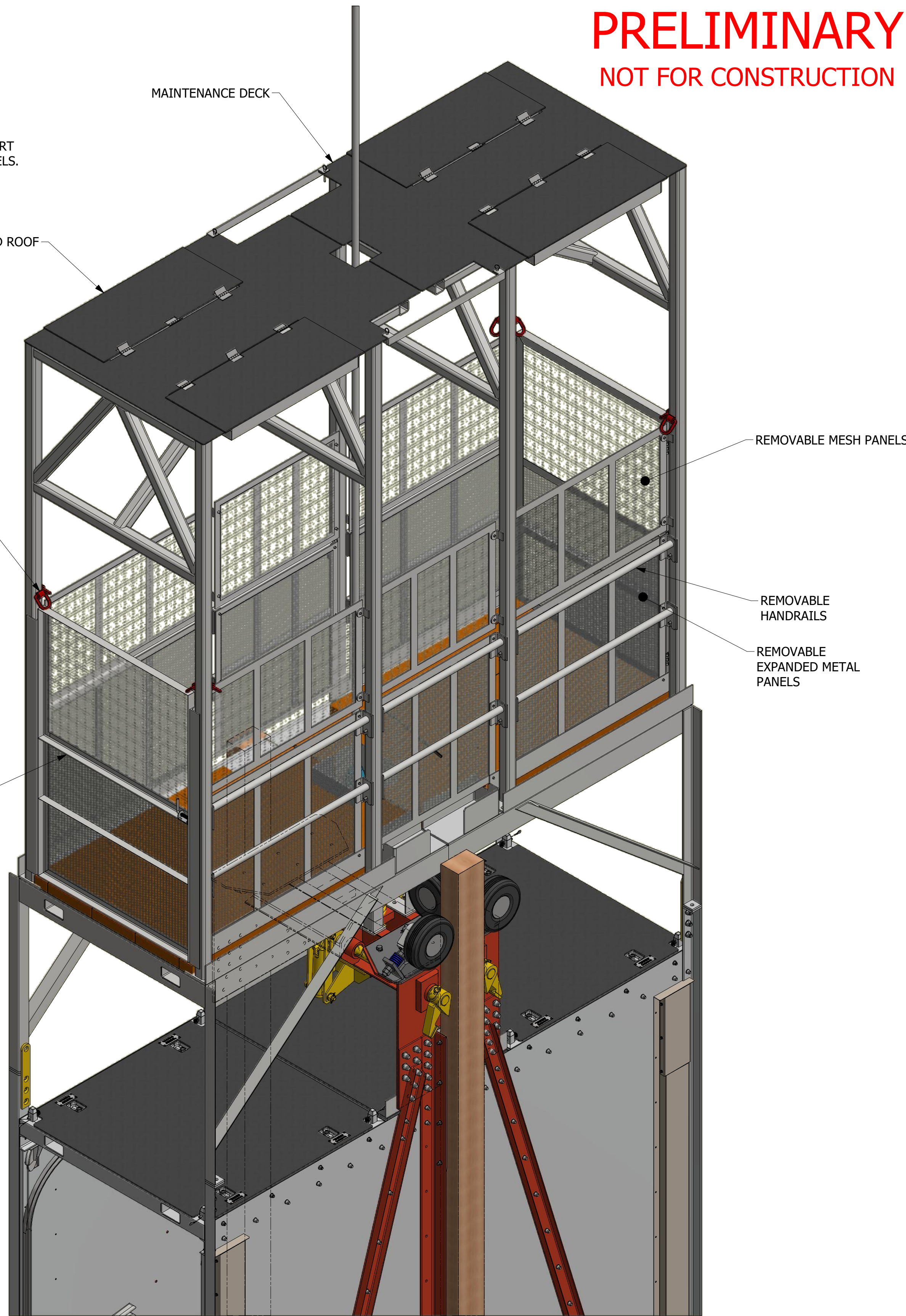
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REV.	DATE	BY	DESCRIPTION	APP.	REV.	DATE	BY	DESCRIPTION	APP.	DWG DATE:
P0	10/15/18	JAD	ISSUED FOR INFORMATION	CLL	P3	11/13/18	JAD	ISSUED FOR 75% DESIGN REVIEW	CLL	10/15/2018
P1	10/25/18	JAD	UPDATE PER DESIGN CHANGE REQUEST	CLL	P4	6/5/19	JAD	ISSUED FOR 90% DESIGN REVIEW	CLL	DESIGN BY: FKC - LS
P2	11/5/18	JAD	UPDATES FOR DESIGN CHANGE REQUEST	CLL	P5	8/30/2019	MAK	ISSUED FOR 100% APPROVAL	CLL	DRAWN BY: JAD
									CHECKED BY: CLL	



**MAINTENANCE DECK - ISO VIEW**  
SCALE 3/4" = 1'-0"

CLIENT	FKC - LAKESHORE EVANSVILLE, IN MAN / MATERIAL CAGE	TITLE	ROSS MAN / MATERIAL CAGE MAINTENANCE DECK ARRANGEMENT	FILE: 8018195-320-02.idw
DRAWING STATUS: PRELIMINARY	ORIGINAL SIZE: 22 x 34	DRAWING NO.	8018195-320-02	REV. P5
		DO NOT SCALE DRAWING	SHEET: 1 OF 3	

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REV.	DATE	BY	DESCRIPTION	APP.	REV.	DATE	BY	DESCRIPTION	APP.	DWG DATE:
P0	10/25/18	JAD	ADD SHEET FOR BRATTICE PANEL HATCH	CLL	P3	6/6/19	JAD	ISSUED FOR 90% DESIGN REVIEW	CLL	10/15/2018
P1	11/5/18	JAD	ADD SHEET FOR BRATTICE PANEL HATCH	CLL	P4	8/30/2019	MAK	ISSUED FOR 100% APPROVAL	CLL	DESIGN BY: FKC - LS
P2	11/13/18	JAD	ISSUED FOR 75% DESIGN REVIEW	CLL					CLL	DRAWN BY: JAD
										CHECKED BY: CLL



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DRAWING STATUS:	PRELIMINARY	ORIGINAL SIZE:	22 x 34	DO NOT SCALE DRAWING	SHEET: 2	OF 3

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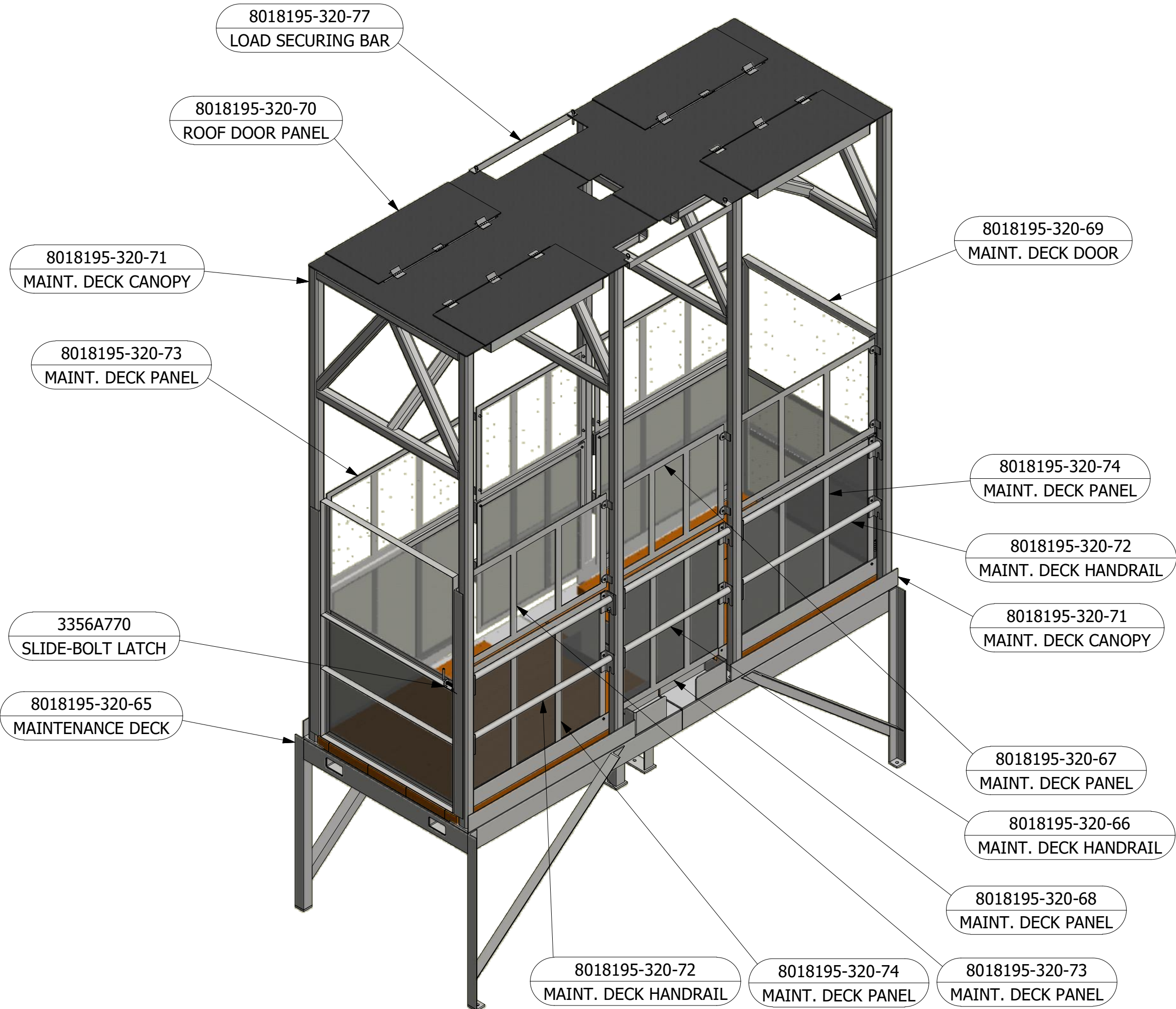
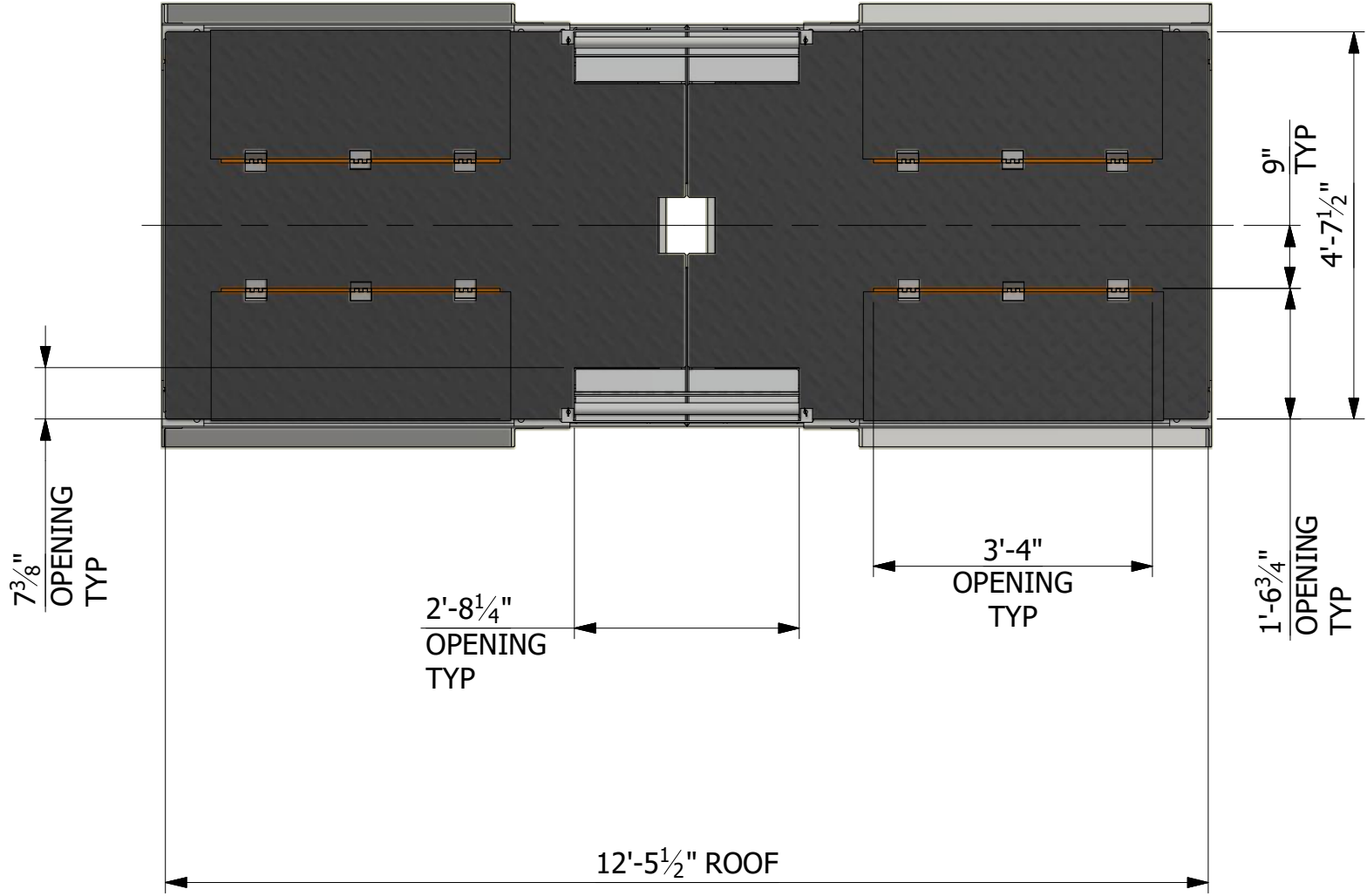
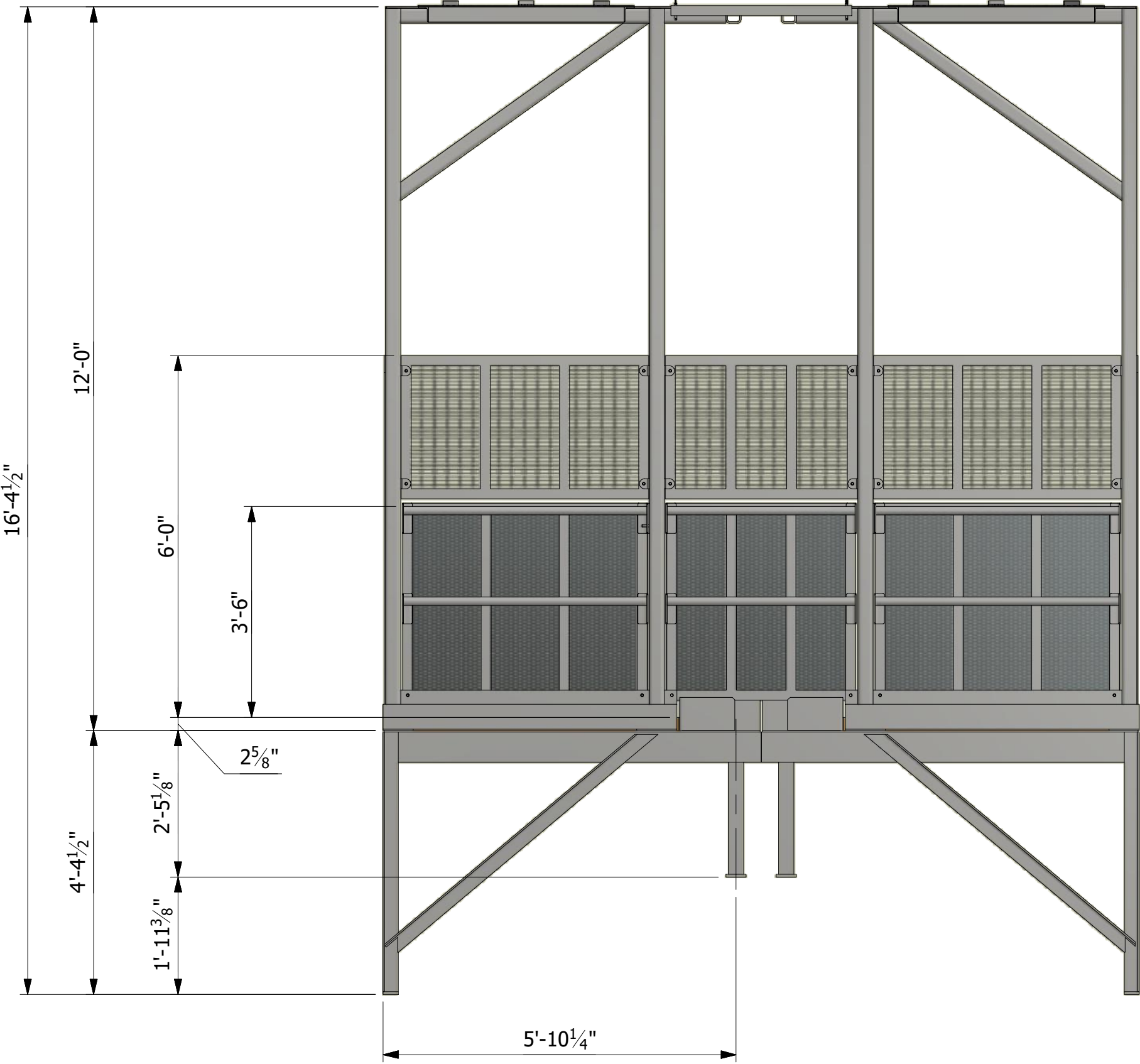
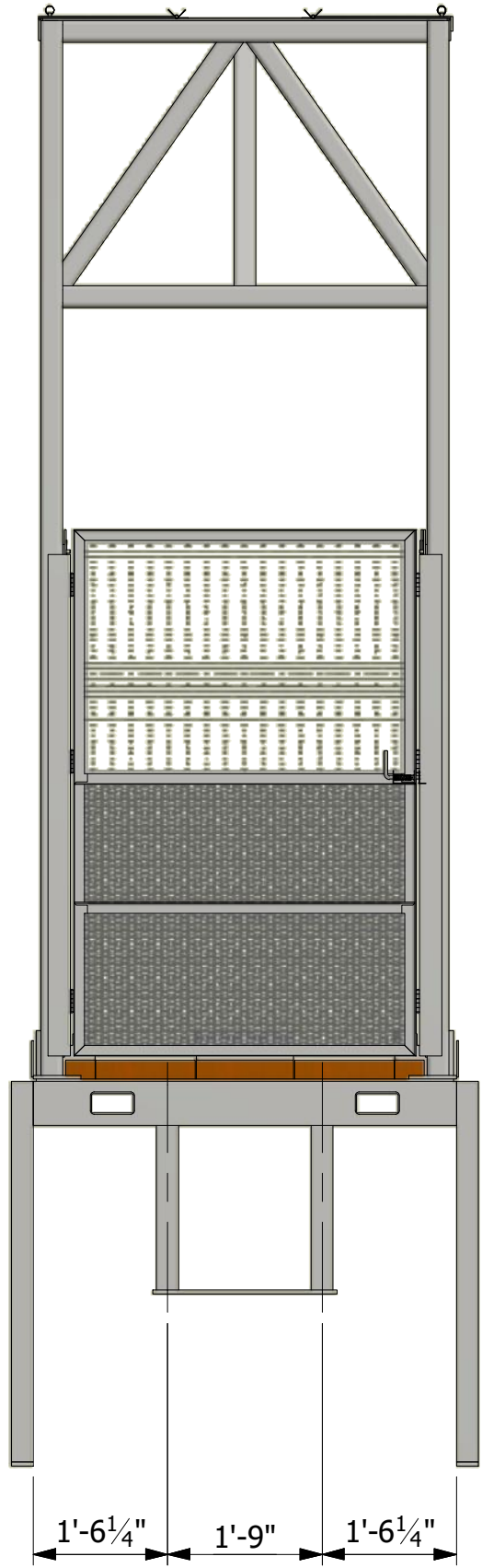
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P0	8/30/2019	MAK	ISSUED FOR 100% APPROVAL	CLL						DESIGN BY:	FKC - LS
										DRAWN BY:	JAD
										CHECKED BY:	CLL

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DATE: \_\_\_\_\_

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PART# 8018195-320-02  
MAINTENANCE DECK  
1/2" = 1'-0"



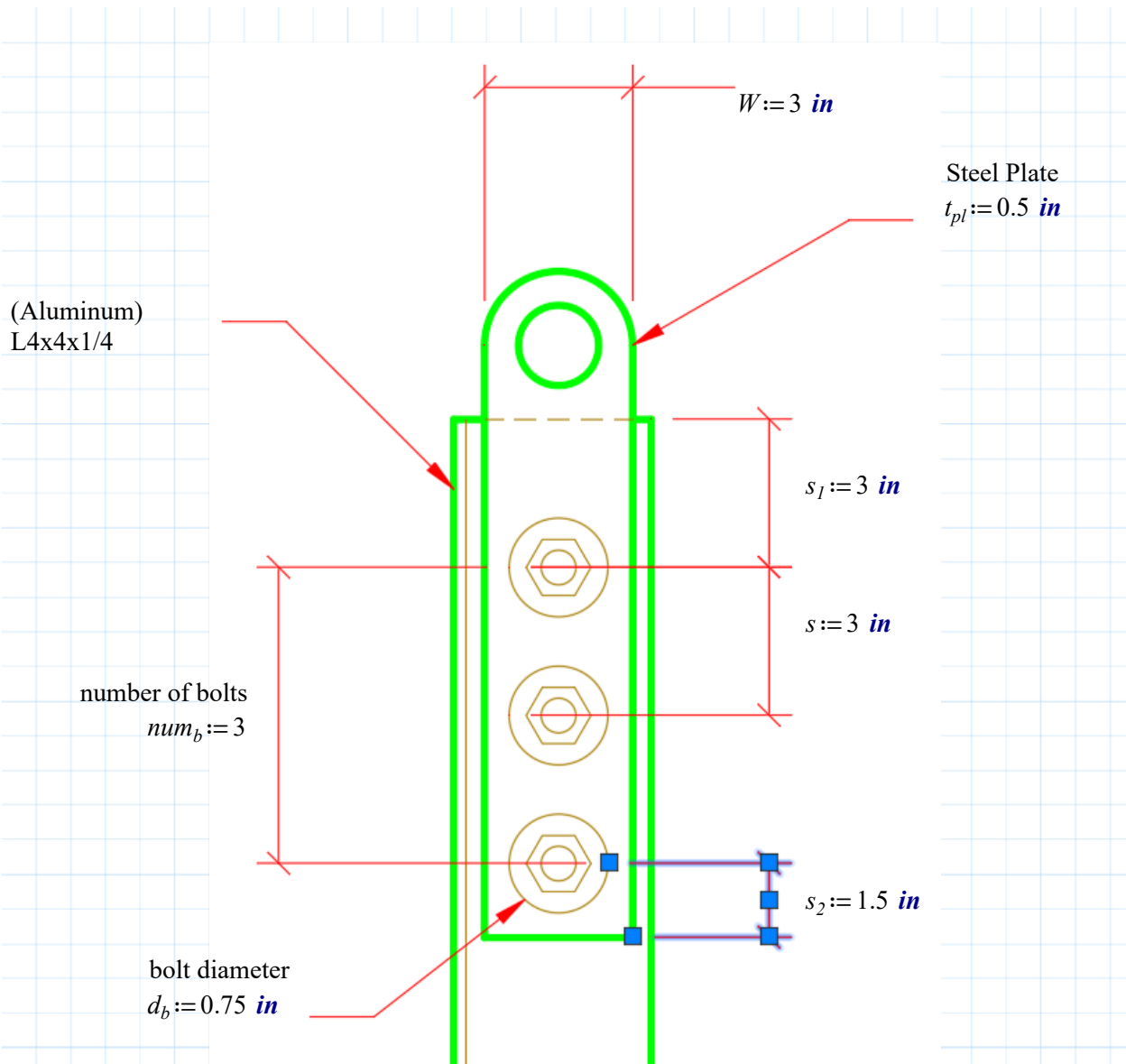
MAINTENANCE DECK  
ISO VIEW

PRELIMINARY  
NOT FOR CONSTRUCTION

						TOTAL WEIGHT (LB)	
						5939	
14	1	3356A770	SLIDE-BOLT LATCH	3" WD x 1 3/4" HI, 1/2" DIA. BOLT	McMASTER CARR	0	
13	6	1609A220	HINGE	3" x 1 1/2" LEAF, .012 THK	McMASTER CARR	3	
12	4	8018195-320-74	MAINT. DECK PANEL	SEE ASSEMBLY DRAWING		175	
11	4	8018195-320-73	MAINT. DECK PANEL	SEE ASSEMBLY DRAWING		232	
10	8	8018195-320-72	MAINT. DECK HANDRAIL	SEE ASSEMBLY DRAWING		105	
9	2	8018195-320-78	LOAD SECURING BAR	SEE ASSEMBLY DRAWING		22	
8	2	8018195-320-77	LOAD SECURING BAR	SEE ASSEMBLY DRAWING		11	
7	2	8018195-320-71	MAINT. DECK CANOPY	SEE ASSEMBLY DRAWING		2687	
6	4	8018195-320-70	ROOF DOOR PANEL	SEE ASSEMBLY DRAWING		387	
5	2	8018195-320-69	MAINT. DECK DOOR	SEE ASSEMBLY DRAWING		132	
4	2	8018195-320-68	MAINT. DECK PANEL	SEE ASSEMBLY DRAWING		76	
3	2	8018195-320-67	MAINT. DECK PANEL	SEE ASSEMBLY DRAWING		95	
2	4	8018195-320-66	MAINT. DECK HANDRAIL	SEE ASSEMBLY DRAWING		44	
1	2	8018195-320-65	MAINTENANCE DECK	SEE ASSEMBLY DRAWING		1970	
ITEM	QTY	PART NO.	NOMENCLATURE	DESCRIPTION	MATERIAL	COMMENTS	WT. (LB)
PARTS LIST - REQUIRED FOR ONE ASSEMBLY							

CLIENT		FKC - LAKESHORE EVANSVILLE, IN MAN / MATERIAL CAGE		TITLE		ROSS MAN / MATERIAL CAGE MAINTENANCE DECK ARRANGEMENT		FILE: 8018195-320-02.idw	
DRAWING STATUS: PRELIMINARY		ORIGINAL SIZE: 22 x 34		DRAWING NO. 8018195-320-02		REV. P0		DO NOT SCALE DRAWING	
						SHEET: 3 OF 3			



Steel Bolts

ASTM A193 Grade B8

Tensile Stress

$$F_{u_b} := 70 \text{ ksi}$$

Yield Stress

$$F_{y_b} := 30 \text{ ksi}$$

Structural Steel

$$F_{y_s} := 36 \text{ ksi}$$

$$F_{u_s} := 50 \text{ ksi}$$

Aluminum

$$F_{y_a} := 35 \text{ ksi}$$

$$F_{u_a} := 38 \text{ ksi}$$

## Failure Modes in Steel Plate

1. Bearing Strength at Bolt Holes  
J3.10

clear distance between the edge of the hole and the edge of the adjacent hole or edge of material

$$\text{Bolt Row \#1} \quad l_{c\_1} := s_1 - \left( \frac{d_b + \frac{1}{8} \text{ in}}{2} \right) = 2.563 \text{ in}$$

$$\text{Bolt Row \#2+} \quad l_{c\_2} := s - \left( d_b + \frac{1}{8} \text{ in} \right) = 2.125 \text{ in}$$

## Nominal Strength

Bolt Row #1

$$R_{n\_1} := \min \left( 1.2 \cdot l_{c\_1} \cdot t_{pl} \cdot F_{u\_s}, 2.4 \cdot d_b \cdot t_{pl} \cdot F_{u\_s} \right)$$

$$R_{n\_1} = 45 \text{ kip}$$

Bolt Row #2+

$$R_{n\_2} := \min \left( 1.2 \cdot l_{c\_2} \cdot t_{pl} \cdot F_{u\_s}, 2.4 \cdot d_b \cdot t_{pl} \cdot F_{u\_s} \right)$$

$$R_{n\_2} = 45 \text{ kip}$$

## Design Strength

$$R_{n1} := 1 \cdot R_{n\_1} + (num_b - 1) \cdot R_{n\_2} = 135 \text{ kip}$$

## 2. Rupture Strength of Plate J4.1

Gross Area

$$A_g := W \cdot t_{pl} = 1.5 \text{ in}^2$$

Effective Net Area

$$A_e := A_g - \left( d_b + \frac{1}{8} \cdot \text{in} \right) \cdot t_{pl}$$

$$A_e = 1.063 \text{ in}^2$$

Nominal Strength  
Tensile Yielding

$$R_{n_y} := A_g \cdot F_{y_s} = 54 \text{ kip}$$

Nominal Strength  
Tensile Rupture

$$R_{n_r} := F_{u_s} \cdot A_e = 53 \text{ kip}$$

Design Strength

$$R_{n2} := \min(R_{n_y}, R_{n_r}) = 53.1 \text{ kip}$$

## 3. Block Shear of the Plate

J4.1 N/A

## 4. Failure Modes in Steel Bolts

nominal unthreaded body area  
of bolt

$$A_b := \frac{\pi \cdot d_b^2}{4} = 0.442 \text{ in}^2$$

nominal shear stress

$$F_{nv} := 0.6 \cdot F_{u\_b} = 42 \text{ ksi}$$

nominal strength of single bolt

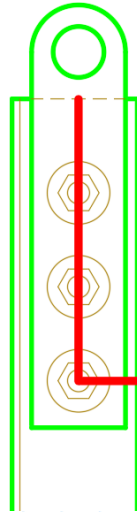
$$R_n := F_{nv} \cdot A_b = 18.555 \text{ kip}$$

Allowable Strength of entire  
bolt group

$$R_{n4} := num_b \cdot R_n = 55.7 \text{ kip}$$

## Failure Modes in Aluminum

### 5. Block Shear in Aluminum Member



thickness of Angle

$$t_l := \frac{1}{4} \text{ in}$$

gross area in shear

$$A_{gv} := (s_l + (num_b - 1) \cdot s) \cdot t_l = 2.3 \text{ in}^2$$

net area in shear

$$A_{nv} := A_{gv} - \left( 2.5 \cdot \left( d_b + \frac{1}{8} \text{ in} \right) \right) \cdot t_l = 1.7 \text{ in}^2$$

gross area in tension

$$A_{gt} := \left( \frac{W}{2} \right) \cdot t_l = 0.375 \text{ in}^2$$

net area in tension

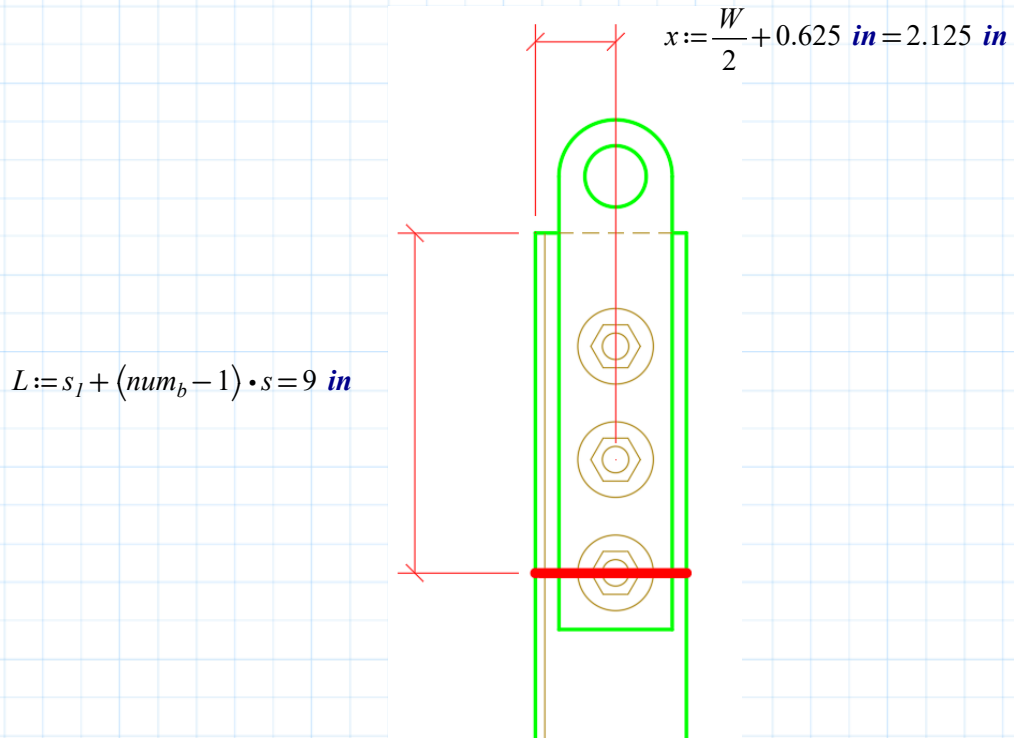
$$A_{nt} := A_{gt} - \left( \frac{1}{2} \cdot \left( d_b + \frac{1}{8} \text{ in} \right) \right) \cdot t_l = 0.266 \text{ in}^2$$

allowable force

$$P_{sr} := \begin{cases} \text{if } F_{u\_a} \cdot A_{nt} \geq (0.6 \cdot F_{u\_a}) \cdot A_{nv} \\ \left\| \left( \left( \frac{F_{y\_a}}{\sqrt{3}} \right) \cdot A_{gv} + F_{u\_a} \cdot A_{nt} \right) \right\| \\ \text{else} \\ \left\| (0.6 \cdot F_{u\_a}) \cdot A_{nv} + F_{y\_a} \cdot A_{gt} \right\| \end{cases}$$

$$R_{n5} := P_{sr} = 52 \text{ kip}$$

## 6. Rupture of Aluminum Angle



Gross Area of angle

$$A_g := 1.94 \text{ in}^2$$

Net Area of angle

$$A_n := A_g - \left( d_b + \frac{1}{8} \text{ in} \right) \cdot t_l = 1.721 \text{ in}^2$$

Shear lag Factor

$$U := 1 - \frac{x}{L} = 0.764$$

Effective net area

$$A_e := A_n \cdot U = 1.315 \text{ in}^2$$

Nominal Strength

$$R_{n6} := F_{u_a} \cdot A_e = 50.0 \text{ kip}$$

## 7. Bolt bearing in Aluminum Member

edge distance

$$\text{bolt row \#1} \quad d_{e1} := \min(s_1, 1.5 \cdot d_b) = 1.125 \text{ in}$$

$$\text{bolt row \#2+} \quad d_{e2} := \min\left(s - 1 \cdot \left(d_b + \frac{1}{8} \text{ in}\right), 1.5 \cdot d_b\right)$$

$$d_{e2} = 1.125 \text{ in}$$

Nominal Strength

Bolt Row #1

$$R_{n\_1} := (2 \cdot F_{u\_a} \cdot t_l \cdot d_b) \cdot \left(\frac{d_{e1}}{2 \cdot d_b}\right) = 10.7 \text{ kip}$$

Bolt Row #2

$$R_{n\_2} := (2 \cdot F_{u\_a} \cdot t_l \cdot d_b) \cdot \left(\frac{d_{e2}}{2 \cdot d_b}\right) = 10.7 \text{ kip}$$

Nominal Strength

$$R_{n7} := R_{n\_1} + (num_b - 1) \cdot R_{n\_2} = 32.1 \text{ kip}$$

### Summary of Failure Modes

1. Bearing Strength @ bolt holes (steel plate)	$R_{n1} = 135 \text{ kip}$
2. Rupture (steel plate)	$R_{n2} = 53.1 \text{ kip}$
3. Block shear (steel plate)	N/A
4. Shear in bolts	$R_{n4} = 55.7 \text{ kip}$
5. Block Shear (Aluminum Member)	$R_{n5} = 52 \text{ kip}$
6. Rupture (Aluminum Member)	$R_{n6} = 50 \text{ kip}$
7. Bolt bearing (Aluminum Member)	$R_{n7} = 32.1 \text{ kip}$

### Governing Failure Mode

Safety Factor

$$\Omega := 5$$

$gov = \text{"7. Bolt Bearing (aluminum member)"}$

$$R_n := \min(R_{n1}, R_{n2}, R_{n4}, R_{n5}, R_{n6}, R_{n7}) = 32.1 \text{ kip}$$

$$R_n := \frac{R_n}{\Omega} = 6.413 \text{ kip}$$

Design Load on Lug

$$P := 4000 \text{ lbf}$$

### Comparison of Demand to Capacity

$$\frac{P}{R_n} = 0.624$$

$$\begin{array}{l|l} \text{if } P \leq R_n & = \text{"GOOD! :)"} \\ \parallel \text{"GOOD! :)"} & \\ \text{else} & \\ \parallel \text{"NO GOOD :("} & \end{array}$$

## Beam in Cage Floor (Supporting Main underlung load)\_

Aluminum Alloy 6061-T6, T6510

ultimate tensile strength  $F_{tu} := 38 \text{ ksi}$

Yield tensile strength  $F_{ty} := 35 \text{ ksi}$

Compression yield strength  $F_{cy} := 35 \text{ ksi}$

Shear yield strength  $F_{su} := 24 \text{ ksi}$

compressive Mod of Elasticity  $E := 10100 \text{ ksi}$

deflection Mod of Elasticity  $E_d := E - 100 \text{ ksi} = 10000 \text{ ksi}$

Safety Factor on Yield Stress  $n_y := 10$

Allowable Stress in Tension Flanges  $F_{at} := \frac{F_{ty}}{n_y} = 3.5 \text{ ksi}$

Allowable Stress in Compression Flanges  $F_{ac} := \frac{F_{cy}}{n_y} = 3.5 \text{ ksi}$

Beam Length  $l := 57.25 \text{ in}$

## Beam Information

Beam Size                      **2 C12x12.1**

Dimensions/Geometry

depth                       $d := 12 \text{ in}$

width                       $b := 3.292 \text{ in}$

Flange Tip  
Thickness                       $t_f := 0.24 \text{ in}$

Ave Flange  
Thickness                       $t := 0.502 \text{ in}$

Web  
Thickness                       $t_w := 0.673 \text{ in}$

Area                       $A := 2 \cdot 10.3 \text{ in}^2$

depth                       $d := 12 \text{ in}$

## Properties (of back to back channels)

## Strong Axis

Mod of Elasticity                       $I_x := 2 \cdot 180 \text{ in}^4$

Section Modulus                       $S_x := 2 \cdot 29.9 \text{ in}^3$

Force on the Lug

$$P := 13500 \text{ lbf}$$

Weight of floor

$$w_d := 20 \text{ psf} \cdot 18 \text{ in} = 30 \text{ plf}$$

Maximum Moment

$$M_{max} := \frac{(w_d) \cdot l^2}{8} \downarrow = 16.2 \text{ kip} \cdot \text{ft} + \frac{(P) \cdot l}{4}$$

Maximum Deflection

$$\Delta_{max} := \frac{5 \cdot (w_d) \cdot l^4}{384 \cdot E_d \cdot I_x} \downarrow = 0.015 \text{ in} + \frac{(P) \cdot l^3}{48 \cdot E_d \cdot I_x}$$

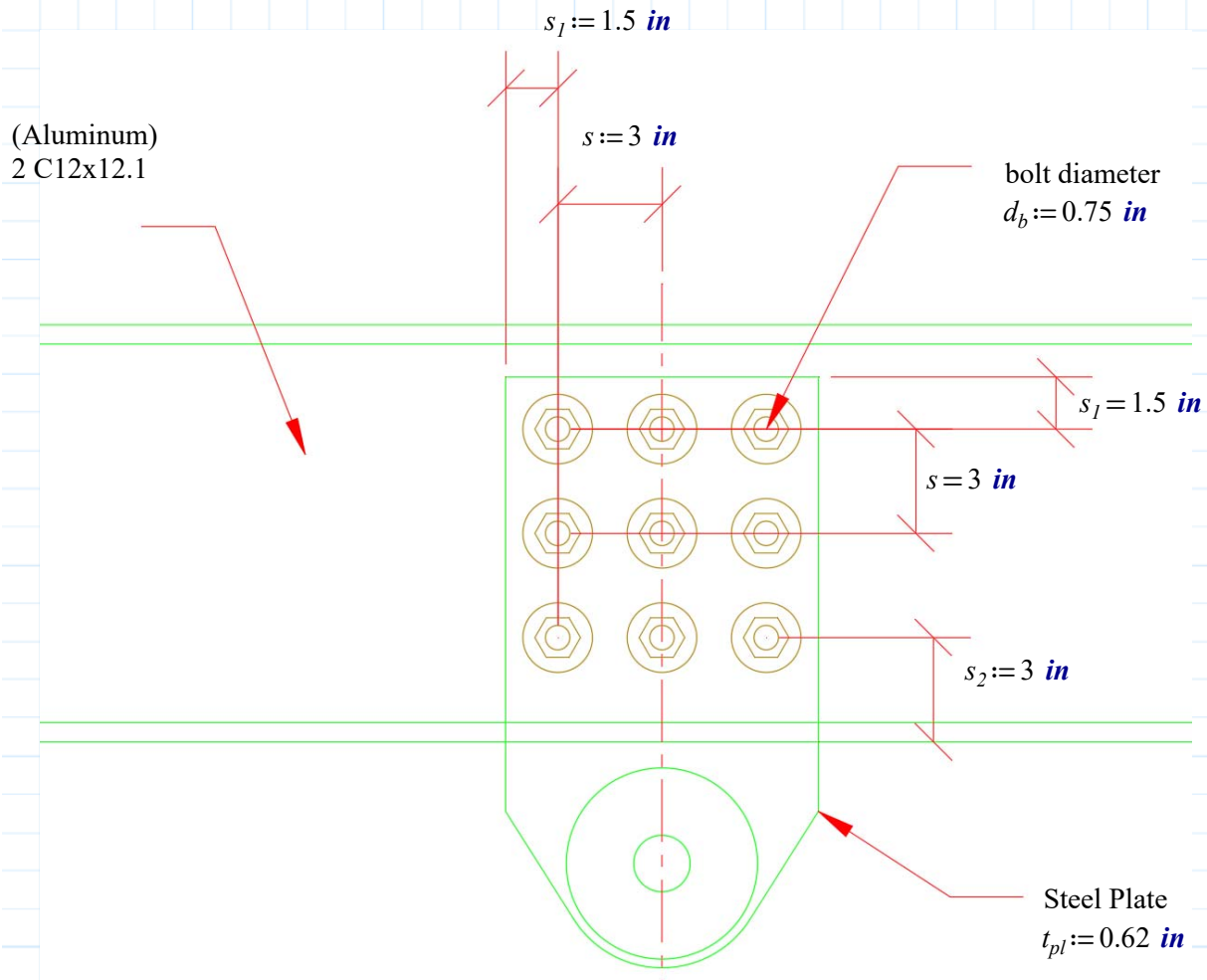
Maximum Stress

$$f_s := \frac{M_{max}}{S_x} = 3.248 \text{ ksi}$$

Maximum Stress  
Allowable Stress

$$\frac{f_s}{F_{at}} = 0.928$$

$$\begin{array}{l|l} \text{if } \frac{f_s}{F_{at}} \leq 1.0 & = \text{"GOOD"} \\ \parallel \text{"GOOD"} & \\ \text{else} & \\ \parallel \text{"NO GOOD"} & \end{array}$$

Steel Bolts

ASTM A193 Grade B8

Tensile Stress

$$F_{u_b} := 70 \text{ ksi}$$

Yield Stress

$$F_{y_b} := 30 \text{ ksi}$$

Structural Steel

$$F_{y_s} := 36 \text{ ksi}$$

$$F_{u_s} := 50 \text{ ksi}$$

Aluminum

$$F_{y_a} := 35 \text{ ksi}$$

$$F_{u_a} := 38 \text{ ksi}$$

## Failure Modes in Steel Plate

1. Bearing Strength at Bolt Holes  
J3.10

clear distance between the edge of the hole and the edge of the adjacent hole or edge of material

$$\text{Bolt Row \#1} \quad l_{c\_1} := s_2 - \left( \frac{d_b + \frac{1}{8} \text{ in}}{2} \right) = 2.563 \text{ in}$$

$$\text{Bolt Row \#2} \quad l_{c\_2} := s - \left( d_b + \frac{1}{8} \text{ in} \right) = 2.125 \text{ in}$$

$$\text{Bolt Row \#3} \quad l_{c\_3} := s - \left( d_b + \frac{1}{8} \text{ in} \right) = 2.125 \text{ in}$$

## Nominal Strength

Bolt Row #1

$$R_{n\_1} := \min \left( 1.2 \cdot l_{c\_1} \cdot t_{pl} \cdot F_{u\_s}, 2.4 \cdot d_b \cdot t_{pl} \cdot F_{u\_s} \right)$$

$$R_{n\_1} = 55.8 \text{ kip}$$

Bolt Row #2

$$R_{n\_2} := \min \left( 1.2 \cdot l_{c\_2} \cdot t_{pl} \cdot F_{u\_s}, 2.4 \cdot d_b \cdot t_{pl} \cdot F_{u\_s} \right)$$

$$R_{n\_2} = 55.8 \text{ kip}$$

Bolt Row #3

$$R_{n\_3} := \min \left( 1.2 \cdot l_{c\_3} \cdot t_{pl} \cdot F_{u\_s}, 2.4 \cdot d_b \cdot t_{pl} \cdot F_{u\_s} \right)$$

$$R_{n\_3} = 55.8 \text{ kip}$$

## Design Strength

$$R_{n1} := 3 \cdot R_{n\_1} + 3 \cdot R_{n\_2} + 3 \cdot R_{n\_3} = 502.2 \text{ kip}$$

## 2. Rupture Strength of Plate J4.1

Gross Area

$$A_g := (2 \cdot s_l + 2 \cdot s) \cdot t_{pl} = 5.58 \text{ in}^2$$

Effective Net Area

$$A_e := \left( (2 \cdot s_l + 2 \cdot s) - 3 \cdot \left( d_b + \frac{1}{8} \text{ in} \right) \right) \cdot t_{pl}$$

$$A_e = 3.953 \text{ in}^2$$

Nominal Strength  
Tensile Yielding

$$R_{n_y} := A_g \cdot F_{y_s} = 200.88 \text{ kip}$$

Nominal Strength  
Tensile Rupture

$$R_{n_r} := F_{u_s} \cdot A_e = 198 \text{ kip}$$

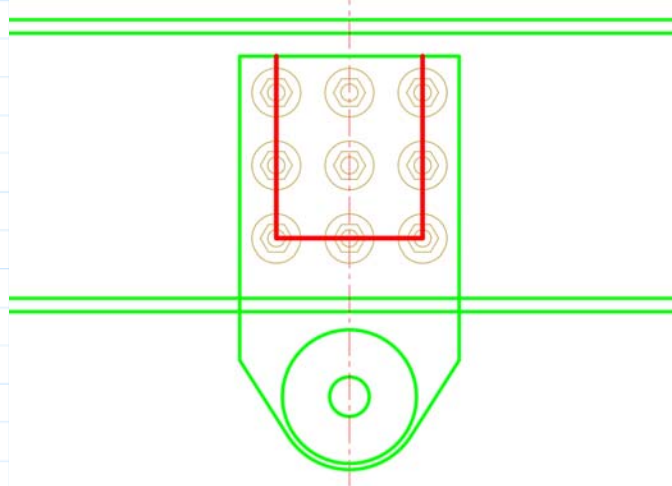
Design Strength

$$R_{n2} := \min(R_{n_y}, R_{n_r}) = 197.6 \text{ kip}$$

## 3. Block Shear of the Plate

J4.1

Situation 1



gross area subject to shear

$$A_{gv} := 2 \cdot (s_l + 2 \cdot s) \cdot t_{pl}$$

$$A_{gv} = 9.3 \text{ in}^2$$

net area subject to shear

$$A_{nv} := 2 \cdot \left( (s_l + 2 \cdot s) - 2.5 \cdot \left( d_b + \frac{1}{8} \text{ in} \right) \right) \cdot t_{pl}$$

$$A_{nv} = 6.588 \text{ in}^2$$

net area subject to tension

$$A_{nt} := \left( 2 \cdot s - 2 \cdot \left( d_b + \frac{1}{8} \text{ in} \right) \right) \cdot t_{pl}$$

$$A_{nt} = 2.635 \text{ in}^2$$

uniform stress factor

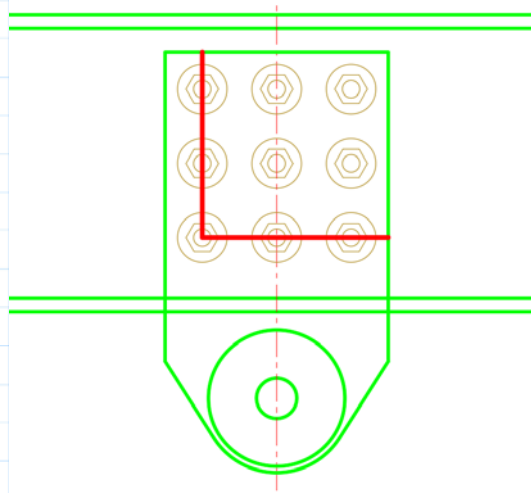
$$U_{bs} := 1.0$$

Nominal Strength

$$R_{n\_I} := \min \left( 0.60 \cdot F_{u\_s} \cdot A_{nv} + U_{bs} \cdot F_{u\_s} \cdot A_{nt}, 0.60 \cdot F_{y\_s} \cdot A_{gv} + U_{bs} \cdot F_{u\_s} \cdot A_{nt} \right)$$

$$R_{n\_I} = 329 \text{ kip}$$

## Situation 2



gross area subject to shear

$$A_{gv} := (s_l + 2 \cdot s) \cdot t_{pl}$$

$$A_{gv} = 4.65 \text{ in}^2$$

net area subject to shear

$$A_{nv} := \left( (s_l + 2 \cdot s) - 2.5 \cdot \left( d_b + \frac{1}{8} \text{ in} \right) \right) \cdot t_{pl}$$

$$A_{nv} = 3.294 \text{ in}^2$$

net area subject to tension

$$A_{nt} := \left( (2 \cdot s + s_l) - 2.5 \cdot \left( d_b + \frac{1}{8} \text{ in} \right) \right) \cdot t_{pl}$$

$$A_{nt} = 3.294 \text{ in}^2$$

uniform stress factor

$$U_{bs} := 1.0$$

Nominal Strength

$$R_{n\_2} := \min \left( 0.60 \cdot F_{u\_s} \cdot A_{nv} + U_{bs} \cdot F_{u\_s} \cdot A_{nt}, 0.60 \cdot F_{y\_s} \cdot A_{gv} + U_{bs} \cdot F_{u\_s} \cdot A_{nt} \right)$$

$$R_{n\_2} = 264 \text{ kip}$$

Block Shear Design Strength

$$R_{n3} := \min (R_{n\_1}, R_{n\_2}) = 263.5 \text{ kip}$$

## 4. Failure Modes in Steel Bolts

number of bolts

$$n_b := 9$$

nominal unthreaded body area  
of bolt

$$A_b := \frac{\pi \cdot d_b^2}{4} = 0.442 \text{ in}^2$$

nominal shear stress

$$F_{nv} := 0.6 \cdot F_{u\_b} = 42 \text{ ksi}$$

nominal strength of single bolt

$$R_n := F_{nv} \cdot A_b = 18.555 \text{ kip}$$

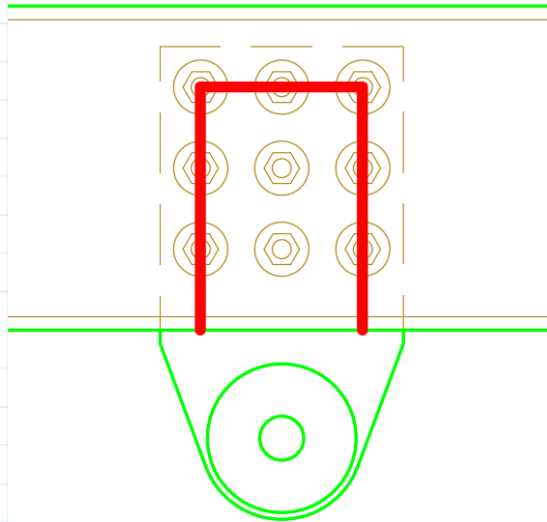
Allowable Strength of entire  
bolt group

$$R_{n4} := n_b \cdot R_n = 167.0 \text{ kip}$$

Failure Modes in Aluminum

## 5. Block Shear in Aluminum Member

C12x12.1 dimensions



flange width

$$b_f := 3.292 \text{ in}$$

flange thickness

$$t_f := 0.502 \text{ in}$$

web thickness

$$t_w := 0.632 \text{ in}$$

gross area in shear

$$A_{gv} := ((2 \cdot s + s_2) \cdot t_w + b_f \cdot t_f) \cdot 4 = 29.4 \text{ in}^2$$

net area in shear

$$A_{nv} := A_{gv} - 4 \cdot \left( 2.5 \cdot \left( d_b + \frac{1}{8} \text{ in} \right) \right) \cdot t_w = 23.8 \text{ in}^2$$

gross area in tension

$$A_{gt} := 2 \cdot (2 \cdot s \cdot t_w) = 7.584 \text{ in}^2$$

net area in tension

$$A_{nt} := A_{gt} - 2 \cdot \left( 2 \cdot \left( d_b + \frac{1}{8} \text{ in} \right) \cdot t_w \right) = 5.372 \text{ in}^2$$

allowable force

$$P_{sr} := \text{if } F_{u\_a} \cdot A_{nt} \geq (0.6 \cdot F_{u\_a}) \cdot A_{nv} \left| \begin{array}{l} \left( \left( \frac{F_{y\_a}}{\sqrt{3}} \right) \cdot A_{gv} + F_{u\_a} \cdot A_{nt} \right) \\ \text{else} \\ \left( (0.6 \cdot F_{u\_a}) \cdot A_{nv} + F_{y\_a} \cdot A_{gt} \right) \end{array} \right|$$

$$R_{ns} := P_{sr} = 808.8 \text{ kip}$$

## 6. Bolt bearing in Aluminum Member

edge distance

$$\text{bolt row \#1} \quad d_{e1} := \min(s_2, 1.5 \cdot d_b) = 1.125 \text{ in}$$

$$\text{bolt row \#2} \quad d_{e2} := \min\left(s - \frac{1}{2} \cdot \left(d_b + \frac{1}{8} \text{ in}\right), 1.5 \cdot d_b\right)$$

$$d_{e2} = 1.125 \text{ in}$$

$$\text{bolt row \#3} \quad d_{e3} := \min\left(s - \frac{1}{2} \cdot \left(d_b + \frac{1}{8} \text{ in}\right), 1.5 \cdot d_b\right)$$

$$d_{e3} = 1.125 \text{ in}$$

Nominal Strength

Bolt Row #1

$$R_{n\_1} := 2 \cdot (2 \cdot F_{u\_a} \cdot t_w \cdot d_b) \cdot \left(\frac{d_{e1}}{2 \cdot d_b}\right) = 54 \text{ kip}$$

Bolt Row #2

$$R_{n\_2} := 2 \cdot (2 \cdot F_{u\_a} \cdot t_w \cdot d_b) \cdot \left(\frac{d_{e2}}{2 \cdot d_b}\right) = 54 \text{ kip}$$

Bolt Row #3

$$R_{n\_3} := 2 \cdot (2 \cdot F_{u\_a} \cdot t_w \cdot d_b) \cdot \left(\frac{d_{e3}}{2 \cdot d_b}\right) = 54 \text{ kip}$$

Design Strength

$$R_{n6} := 3 \cdot R_{n\_1} + 3 \cdot R_{n\_2} + 3 \cdot R_{n\_3} = 486.3 \text{ kip}$$

### Summary of Failure Modes

1. Bearing Strength @ bolt holes (steel plate)	$R_{n1} = 502.2 \text{ kip}$
2. Rupture (steel plate)	$R_{n2} = 197.6 \text{ kip}$
3. Block shear (steel plate)	$R_{n3} = 263.5 \text{ kip}$
4. Shear in bolts	$R_{n4} = 167.0 \text{ kip}$
5. Block Shear (Aluminum Member)	$R_{n5} = 808.8 \text{ kip}$
6. Bolt bearing (Aluminum Member)	$R_{n6} = 486.3 \text{ kip}$

Governing Failure Mode

Safety Factor

$$\Omega := 10$$

gov = "4. Bolt Shear Strength"

$$R_n := \min(R_{n1}, R_{n2}, R_{n3}, R_{n4}, R_{n5}, R_{n6}) = 167.0 \text{ kip}$$

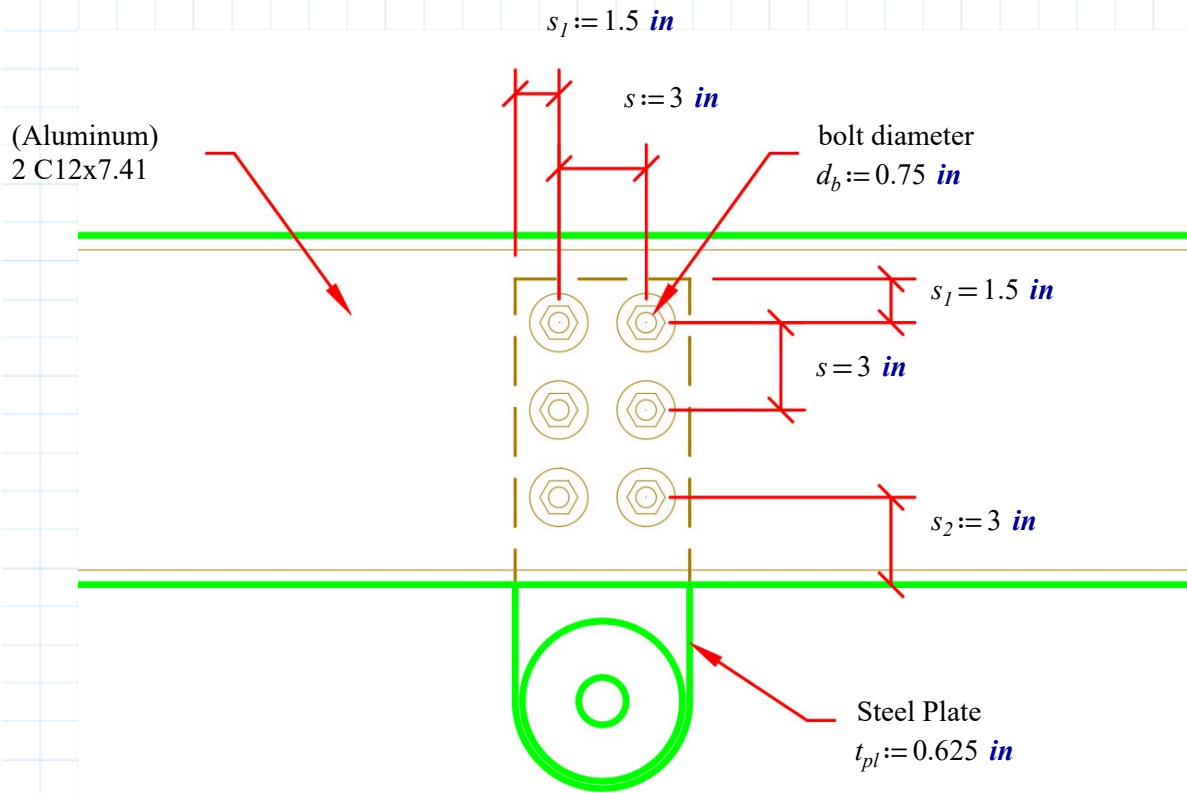
Design Load on Lug

$$P := 13500 \text{ lbf}$$

### Determination of Safety Factor

$$\Omega_{actual} := \frac{R_n}{P} = 12.37$$

if $\frac{R_n}{P} \geq 10.0$    "GOOD! :)" else    "NO GOOD :("	= "GOOD! :)"
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Steel Bolts

ASTM A193 Grade B8

Tensile Stress

$$F_{u\_b} := 70 \text{ ksi}$$

Yield Stress

$$F_{y\_b} := 30 \text{ ksi}$$

Structural Steel

$$F_{y\_s} := 36 \text{ ksi}$$

$$F_{u\_s} := 50 \text{ ksi}$$

Aluminum

$$F_{y\_a} := 35 \text{ ksi}$$

$$F_{u\_a} := 38 \text{ ksi}$$

## Failure Modes in Steel Plate

1. Bearing Strength at Bolt Holes  
J3.10

clear distance between the edge of the hole and the edge of the adjacent hole or edge of material

$$\text{Bolt Row \#1} \quad l_{c\_1} := s_2 - \left( \frac{d_b + \frac{1}{8} \text{ in}}{2} \right) = 2.563 \text{ in}$$

$$\text{Bolt Row \#2} \quad l_{c\_2} := s - \left( d_b + \frac{1}{8} \text{ in} \right) = 2.125 \text{ in}$$

$$\text{Bolt Row \#3} \quad l_{c\_3} := s - \left( d_b + \frac{1}{8} \text{ in} \right) = 2.125 \text{ in}$$

## Nominal Strength

Bolt Row #1

$$R_{n\_1} := \min \left( 1.2 \cdot l_{c\_1} \cdot t_{pl} \cdot F_{u\_s}, 2.4 \cdot d_b \cdot t_{pl} \cdot F_{u\_s} \right)$$

$$R_{n\_1} = 56.25 \text{ kip}$$

Bolt Row #2

$$R_{n\_2} := \min \left( 1.2 \cdot l_{c\_2} \cdot t_{pl} \cdot F_{u\_s}, 2.4 \cdot d_b \cdot t_{pl} \cdot F_{u\_s} \right)$$

$$R_{n\_2} = 56.25 \text{ kip}$$

Bolt Row #3

$$R_{n\_3} := \min \left( 1.2 \cdot l_{c\_3} \cdot t_{pl} \cdot F_{u\_s}, 2.4 \cdot d_b \cdot t_{pl} \cdot F_{u\_s} \right)$$

$$R_{n\_3} = 56.25 \text{ kip}$$

## Design Strength

$$R_{n1} := 2 \cdot R_{n\_1} + 2 \cdot R_{n\_2} + 2 \cdot R_{n\_3} = 337.5 \text{ kip}$$

## 2. Rupture Strength of Plate J4.1

Gross Area

$$A_g := (2 \cdot s_l + 1 \cdot s) \cdot t_{pl} = 3.75 \text{ in}^2$$

Effective Net Area

$$A_e := \left( (2 \cdot s_l + 1 \cdot s) - 2 \cdot \left( d_b + \frac{1}{8} \text{ in} \right) \right) \cdot t_{pl}$$

$$A_e = 2.656 \text{ in}^2$$

Nominal Strength  
Tensile Yielding

$$R_{n_y} := A_g \cdot F_{y_s} = 135 \text{ kip}$$

Nominal Strength  
Tensile Rupture

$$R_{n_r} := F_{u_s} \cdot A_e = 133 \text{ kip}$$

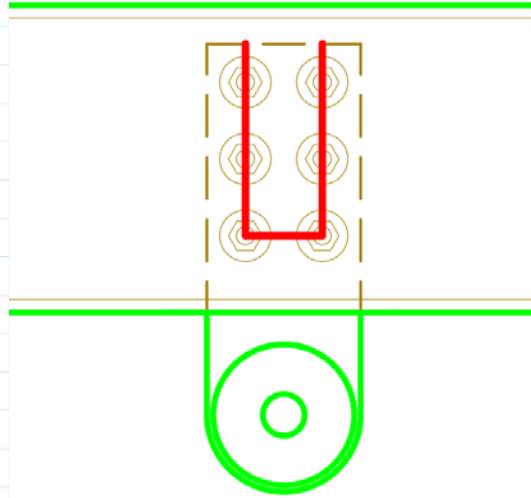
Design Strength

$$R_{n2} := \min(R_{n_y}, R_{n_r}) = 132.8 \text{ kip}$$

## 3. Block Shear of the Plate

J4.1

Situation 1



gross area subject to shear

$$A_{gv} := 2 \cdot (s_l + 2 \cdot s) \cdot t_{pl}$$

$$A_{gv} = 9.375 \text{ in}^2$$

net area subject to shear

$$A_{nv} := 2 \cdot \left( (s_l + 2 \cdot s) - 2.5 \cdot \left( d_b + \frac{1}{8} \text{ in} \right) \right) \cdot t_{pl}$$

$$A_{nv} = 6.641 \text{ in}^2$$

net area subject to tension

$$A_{nt} := \left( 1 \cdot s - 1 \cdot \left( d_b + \frac{1}{8} \text{ in} \right) \right) \cdot t_{pl}$$

$$A_{nt} = 1.328 \text{ in}^2$$

uniform stress factor

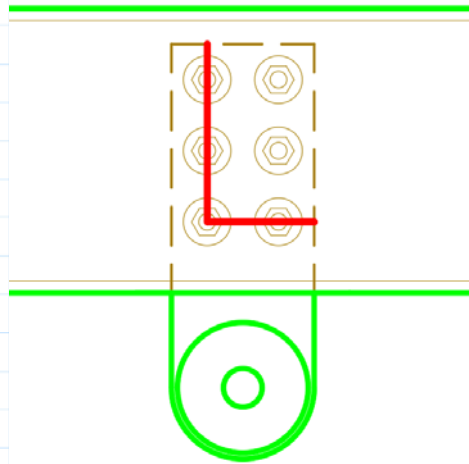
$$U_{bs} := 1.0$$

Nominal Strength

$$R_{n\_I} := \min \left( 0.60 \cdot F_{u\_s} \cdot A_{nv} + U_{bs} \cdot F_{u\_s} \cdot A_{nt}, 0.60 \cdot F_{y\_s} \cdot A_{gv} + U_{bs} \cdot F_{u\_s} \cdot A_{nt} \right)$$

$$R_{n\_I} = 266 \text{ kip}$$

Situation 2



gross area subject to shear

$$A_{gv} := (s_l + 2 \cdot s) \cdot t_{pl}$$

$$A_{gv} = 4.688 \text{ in}^2$$

net area subject to shear

$$A_{nv} := \left( (s_l + 2 \cdot s) - 2.5 \cdot \left( d_b + \frac{1}{8} \text{ in} \right) \right) \cdot t_{pl}$$

$$A_{nv} = 3.32 \text{ in}^2$$

net area subject to tension

$$A_{nt} := \left( (1 \cdot s + s_l) - 1.5 \cdot \left( d_b + \frac{1}{8} \text{ in} \right) \right) \cdot t_{pl}$$

$$A_{nt} = 1.992 \text{ in}^2$$

uniform stress factor

$$U_{bs} := 1.0$$

Nominal Strength

$$R_{n\_2} := \min \left( 0.60 \cdot F_{u\_s} \cdot A_{nv} + U_{bs} \cdot F_{u\_s} \cdot A_{nt}, 0.60 \cdot F_{y\_s} \cdot A_{gv} + U_{bs} \cdot F_{u\_s} \cdot A_{nt} \right)$$

$$R_{n\_2} = 199 \text{ kip}$$

Block Shear Design Strength

$$R_{n3} := \min (R_{n\_1}, R_{n\_2}) = 199.2 \text{ kip}$$

## 4. Failure Modes in Steel Bolts

number of bolts

$$n_b := 6$$

nominal unthreaded body area  
of bolt

$$A_b := \frac{\pi \cdot d_b^2}{4} = 0.442 \text{ in}^2$$

nominal shear stress

$$F_{nv} := 0.6 \cdot F_{u\_b} = 42 \text{ ksi}$$

nominal strength of single bolt

$$R_n := F_{nv} \cdot A_b = 18.555 \text{ kip}$$

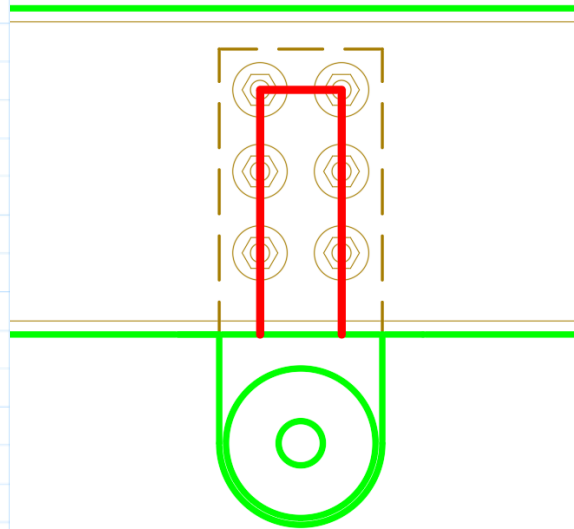
Allowable Strength of entire  
bolt group

$$R_{n4} := n_b \cdot R_n = 111.3 \text{ kip}$$

Failure Modes in Aluminum

## 5. Block Shear in Aluminum Member

C12x7.41 dimensions



flange width

$$b_f := 2.960 \text{ in}$$

flange thickness

$$t_f := 0.502 \text{ in}$$

web thickness

$$t_w := 0.300 \text{ in}$$

gross area in shear

$$A_{gv} := ((2 \cdot s + s_2) \cdot t_w + b_f \cdot t_f) \cdot 4 = 16.7 \text{ in}^2$$

net area in shear

$$A_{nv} := A_{gv} - 4 \cdot \left( 2.5 \cdot \left( d_b + \frac{1}{8} \text{ in} \right) \right) \cdot t_w = 14.1 \text{ in}^2$$

gross area in tension

$$A_{gt} := 2 \cdot (2 \cdot s \cdot t_w) = 3.6 \text{ in}^2$$

net area in tension

$$A_{nt} := A_{gt} - 2 \cdot \left( 1 \cdot \left( d_b + \frac{1}{8} \text{ in} \right) \cdot t_w \right) = 3.075 \text{ in}^2$$

allowable force

$$P_{sr} := \text{if } F_{u\_a} \cdot A_{nt} \geq (0.6 \cdot F_{u\_a}) \cdot A_{nv} \left| \begin{array}{l} \left( \left( \frac{F_{y\_a}}{\sqrt{3}} \right) \cdot A_{gv} + F_{u\_a} \cdot A_{nt} \right) \\ \text{else} \\ \left( (0.6 \cdot F_{u\_a}) \cdot A_{nv} + F_{y\_a} \cdot A_{gt} \right) \end{array} \right|$$

$$R_{ns} := P_{sr} = 447.9 \text{ kip}$$

## 6. Bolt bearing in Aluminum Member

edge distance

$$\text{bolt row \#1} \quad d_{e1} := \min(s_2, 1.5 \cdot d_b) = 1.125 \text{ in}$$

$$\text{bolt row \#2} \quad d_{e2} := \min\left(s - \frac{1}{2} \cdot \left(d_b + \frac{1}{8} \text{ in}\right), 1.5 \cdot d_b\right)$$

$$d_{e2} = 1.125 \text{ in}$$

$$\text{bolt row \#3} \quad d_{e3} := \min\left(s - \frac{1}{2} \cdot \left(d_b + \frac{1}{8} \text{ in}\right), 1.5 \cdot d_b\right)$$

$$d_{e3} = 1.125 \text{ in}$$

Nominal Strength

Bolt Row #1

$$R_{n\_1} := 2 \cdot (2 \cdot F_{u\_a} \cdot t_w \cdot d_b) \cdot \left(\frac{d_{e1}}{2 \cdot d_b}\right) = 25.7 \text{ kip}$$

Bolt Row #2

$$R_{n\_2} := 2 \cdot (2 \cdot F_{u\_a} \cdot t_w \cdot d_b) \cdot \left(\frac{d_{e2}}{2 \cdot d_b}\right) = 25.7 \text{ kip}$$

Bolt Row #3

$$R_{n\_3} := 2 \cdot (2 \cdot F_{u\_a} \cdot t_w \cdot d_b) \cdot \left(\frac{d_{e3}}{2 \cdot d_b}\right) = 25.7 \text{ kip}$$

Design Strength

$$R_{n6} := 2 \cdot R_{n\_1} + 2 \cdot R_{n\_2} + 2 \cdot R_{n\_3} = 153.9 \text{ kip}$$

### Summary of Failure Modes

1. Bearing Strength @ bolt holes (steel plate)	$R_{n1} = 337.5 \text{ kip}$
2. Rupture (steel plate)	$R_{n2} = 132.8 \text{ kip}$
3. Block shear (steel plate)	$R_{n3} = 199.2 \text{ kip}$
4. Shear in bolts	$R_{n4} = 111.3 \text{ kip}$
5. Block Shear (Aluminum Member)	$R_{n5} = 447.9 \text{ kip}$
6. Bolt bearing (Aluminum Member)	$R_{n6} = 153.9 \text{ kip}$

Governing Failure Mode

Safety Factor

$$\Omega := 10$$

gov = "4. Bolt Shear Strength"

$$R_n := \min(R_{n1}, R_{n2}, R_{n3}, R_{n4}, R_{n5}, R_{n6}) = 111.3 \text{ kip}$$

$$R_n := \frac{R_n}{\Omega} = 11.133 \text{ kip}$$

Design Load on Lug

$$P := \frac{13500}{2} \text{ lbf} = 6750 \text{ lbf}$$

### Comparison of Demand to Capacity

$$\frac{P}{R_n} = 0.606$$

if $P \leq R_n$    "GOOD! :)" else    "NO GOOD :("	= "GOOD! :)"
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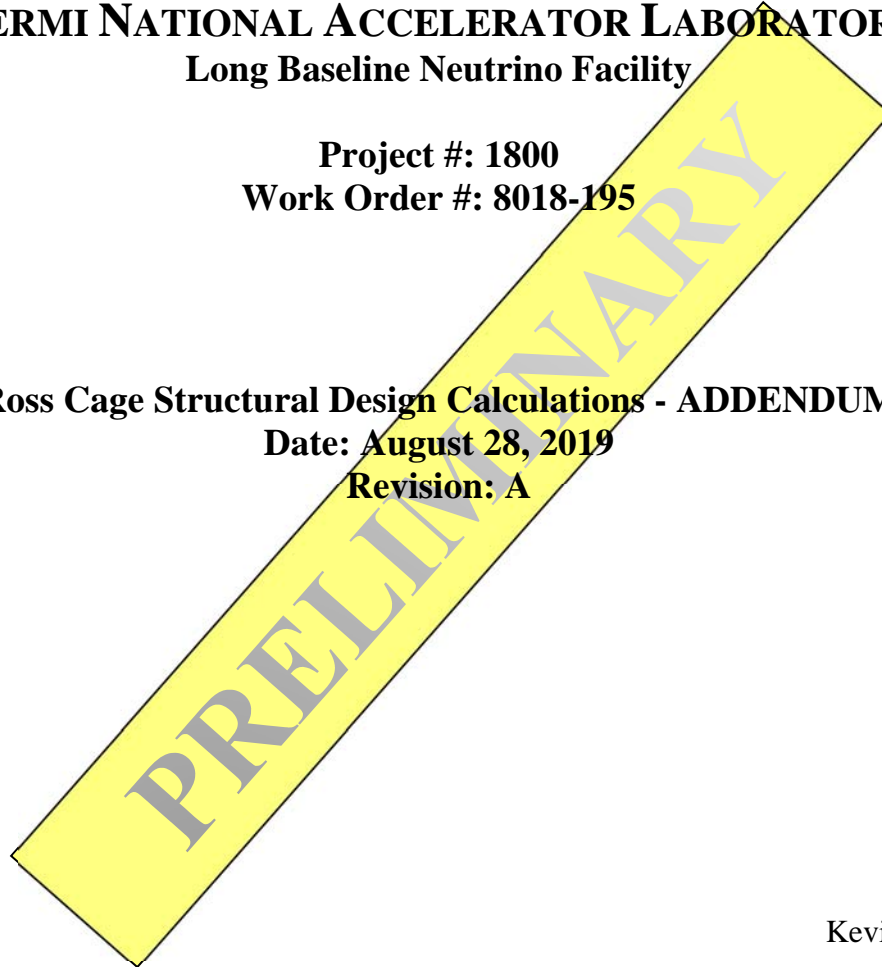
**FERMI NATIONAL ACCELERATOR LABORATORY  
Long Baseline Neutrino Facility**

**Project #: 1800  
Work Order #: 8018-195**

**Ross Cage Structural Design Calculations - ADDENDUM**

**Date: August 28, 2019**

**Revision: A**



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## **1 DESIGN SUMMARY/PURPOSE/OVERVIEW**

### **1.1 PURPOSE OF DOCUMENT**

This document summarizes the calculations made for the Maintenance Deck Platforms. This is a supplemental structure that is attached to the Main Ross Cage only during maintenance activities. It is *not* considered part of the permanent structure

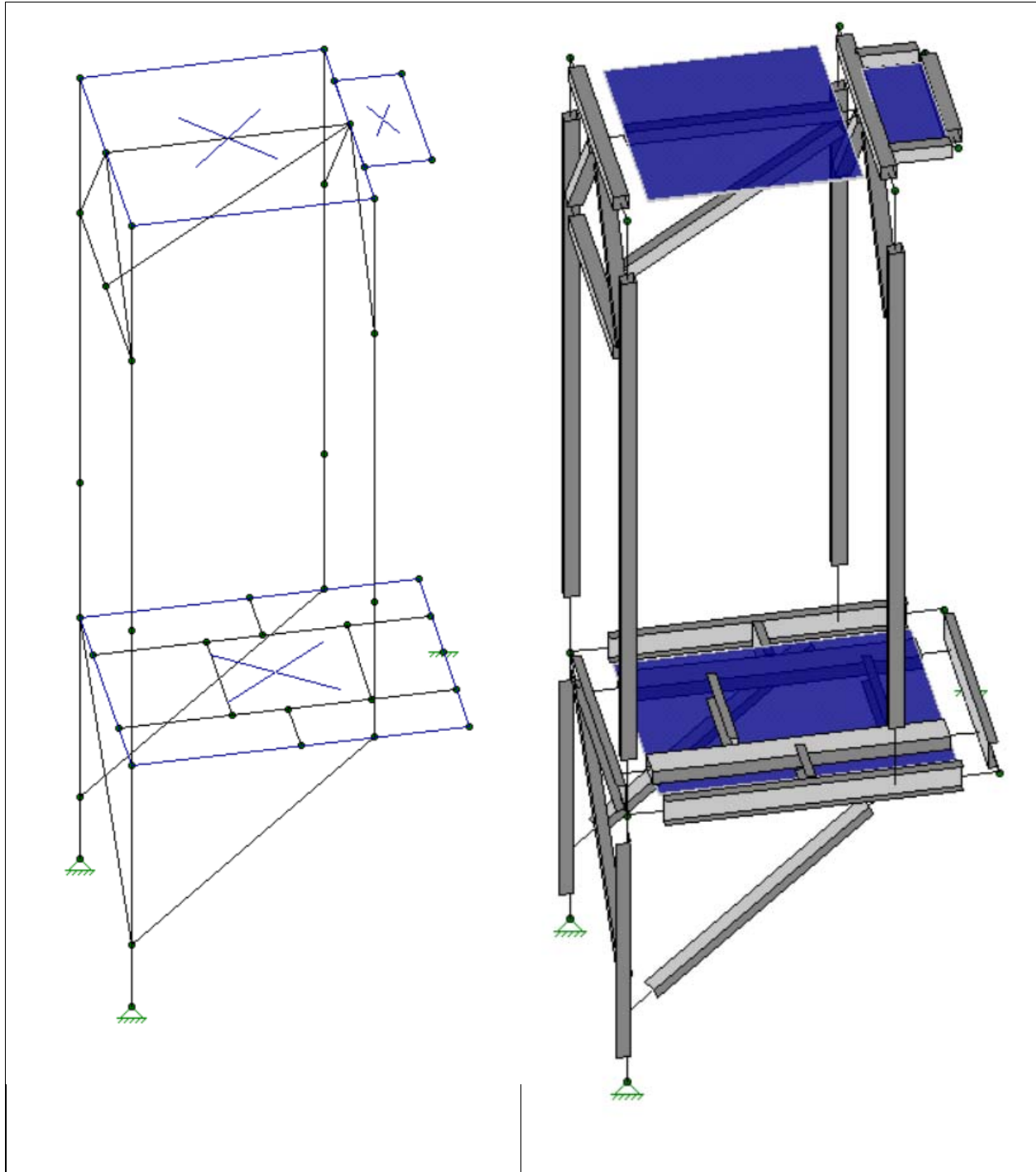
## **2 DESIGN CRITERIA/REFERENCE INFORMATION**

### **2.1 DESIGN CODES**

International Building Code	IBC-2015
Minimum Design Loads for Buildings and Other Structures,	ASCE 7-10
Design Loads on Structures during Construction,	ASCE 37-14
Building Code Requirements for Structural Concrete,	ACI 318-11
Specifications for Structural Steel Buildings,	AISC 360-10
Specifications for Aluminum Structures	AA SAS30-10
Mine Safety and Health Administration Standards & Regulations	MSHA CFR Title 30

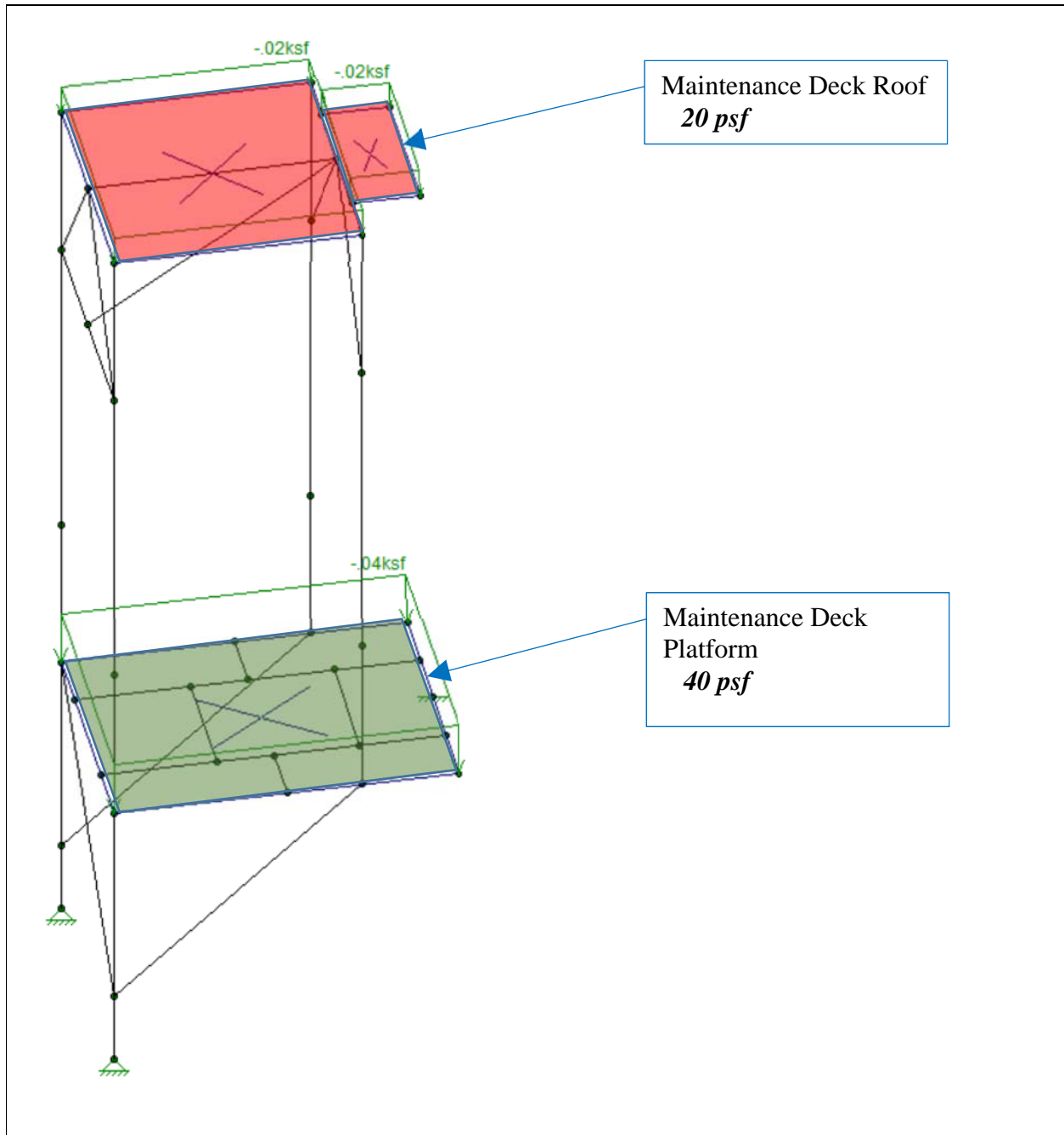
## 2.2 DESIGN LOADS

### Dead Loads

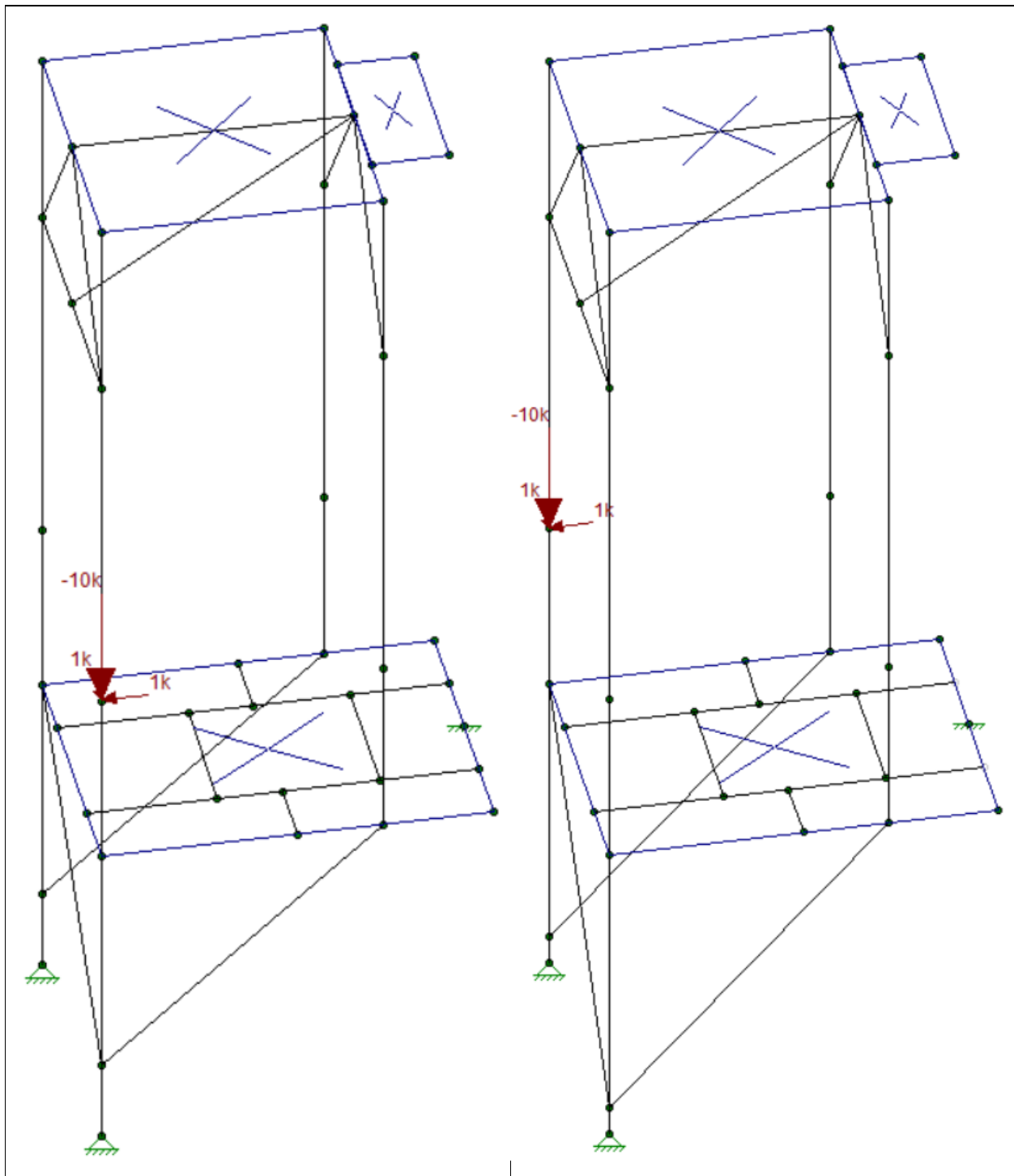


Dead Loads for the structure consist solely of the self-weight of the steel materials (HSS, Angles, Channels, Plates, Etc.) that make up the structure. Note that since the structure is symmetric, only 1 side of the structure is modelled.

## Live Loads – Platform



## Live Loads – Personnel Tie-off Points



There are (2) tie-off points on each corner post, 8 total. We consider the case where (2) tie-off points on a single post are used simultaneously. Each tie-off point supports 5000 lbs (vertical load) and 500 lbs (horizontal in both axis).

**Soil Loads**

N/A

**Snow Loads**

N/A

**Rain Loads**

N/A

**Ice Loads**

N/A

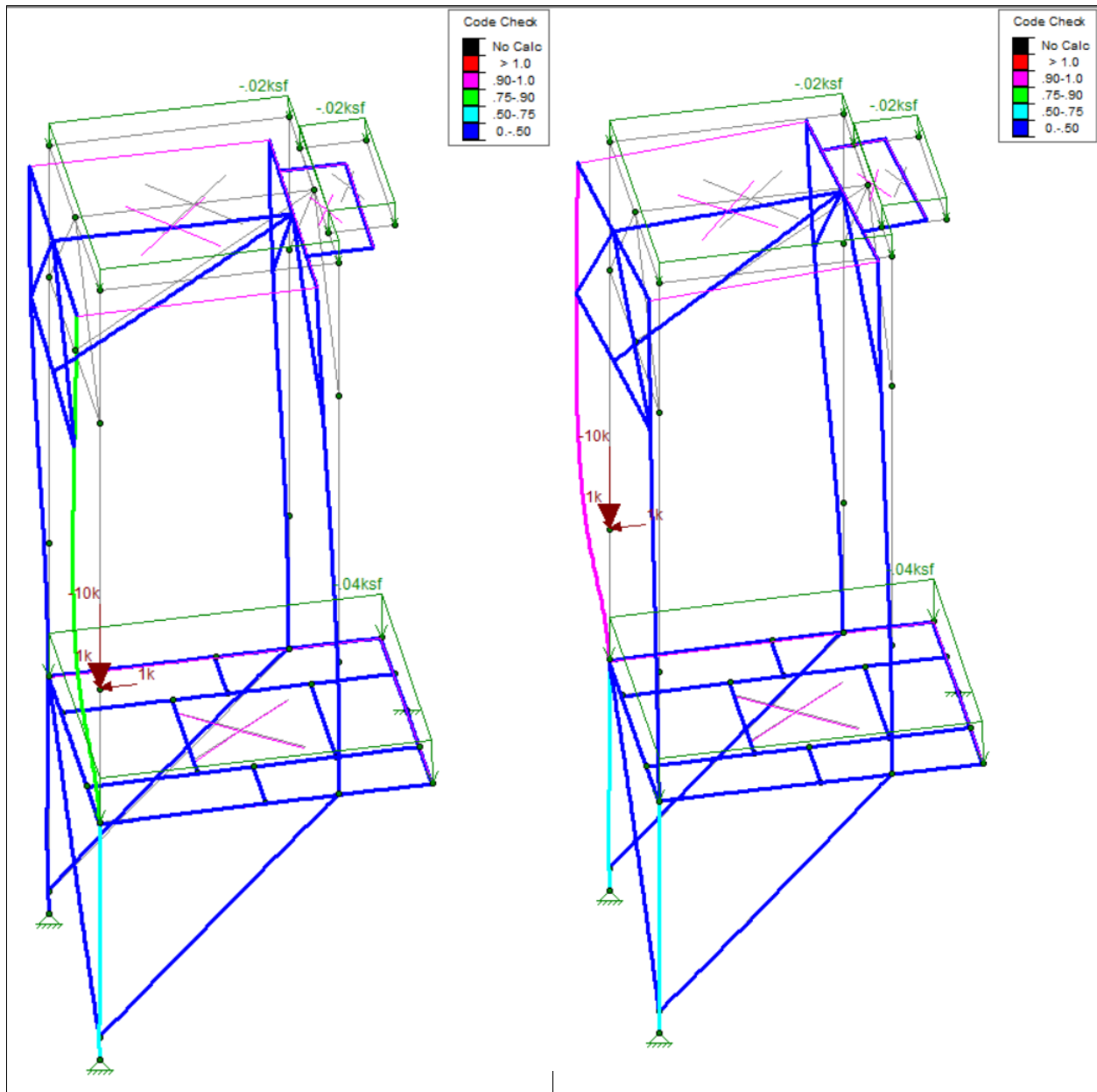
**Seismic Loads**

N/A

**Wind Loads**

N/A

### 3 CALCULATIONS/STRUCTURAL MODEL



Results of the Calculations are shown here for the two tie-off point situations. "Code Check" refers to the ratio of a member's demand to capacity. A ratio less than 1.0 is good.

## **4 CONCLUSIONS**

A structural model was created to analyze the Maintenance Deck structure. This is a supplemental structure attached to the main Ross Cage during maintenance activities.

Unlike the main cage (which is composed of mainly aluminum), the maintenance deck is composed of structural steel rolled shapes. It is bolted into the main deck at the 4 corners, as well as the middle over the main cage transom beam.

The structural model accounts for the dead load of the major structural materials used. It also accounts for a blanket live load over both the work deck below and the roof deck above. Additionally, it accounts for personnel tie-off points on each of the corner posts.

The structural analysis showed the members are adequate for the expected loads.