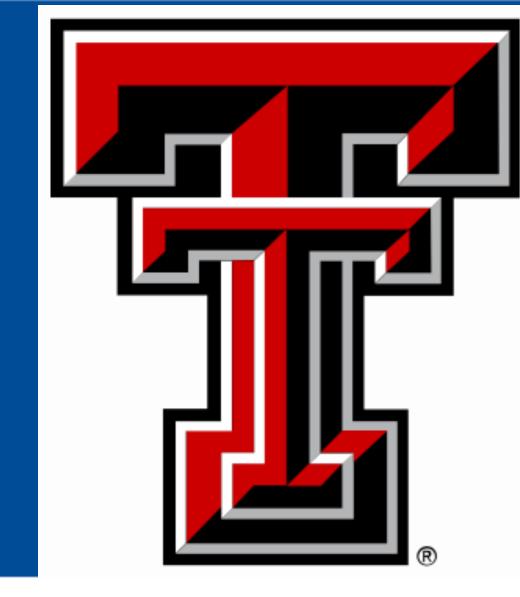


# **Design and Production of the CMS High Granularity Calorimeter Mockup Modules** Kamal Lamichhane, Texas Tech University



## Introduction

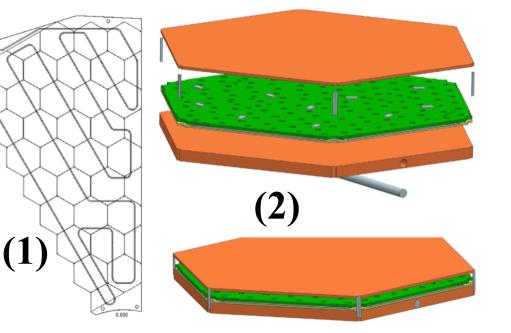
- High luminosity-LHC plans 5 x  $10^{34}$  cm<sup>-2</sup>s<sup>-1</sup> instantaneous luminosity and 3000 fb<sup>-1</sup> integrated luminosity.
- High pile-up conditions ~200 interactions per bunch crossing and high radiation dose: ~2MGy,  $10^{16} n_{eq}/cm^2$ . [ $n_{eq}/cm^2$  denotes the number of 1 MeV equivalent neutrons per  $cm^2$ ]
- Current electromagnetic (E) and hadronic (H) endcap calorimeter cannot stand the radiation dose of HL-LHC & can only sustain ~ 500 fb<sup>-1</sup>.
- Hence, CMS plans to replace the current endcap calorimeters (CE) with HGCAL during LS3 [2024-2026].

# **HGCAL mockup module**

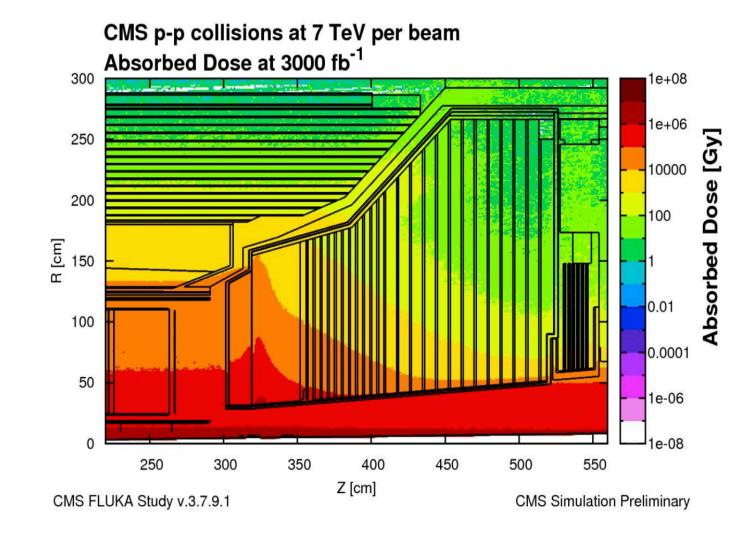
- **Base-plates** (1 mm thick):
  - Provides support for the Si & enables mounting to cassette.  $\bullet$
  - Design: Carbon fiber (H), Copper/Tungsten (E);
    - To have the coefficient of thermal expansion (CTE) as close • to silicon as possible and good thermal conductivity.
  - Investigating other choices also: brass, stainless steel, ceramics.
  - Mock-up test are so far done on brass plates.
- Kapton sheet (110 µm thick):
  - Coated with a thin layer of gold: to provide the HV bias

# **Finite Element Analysis (FEA) Simulation**

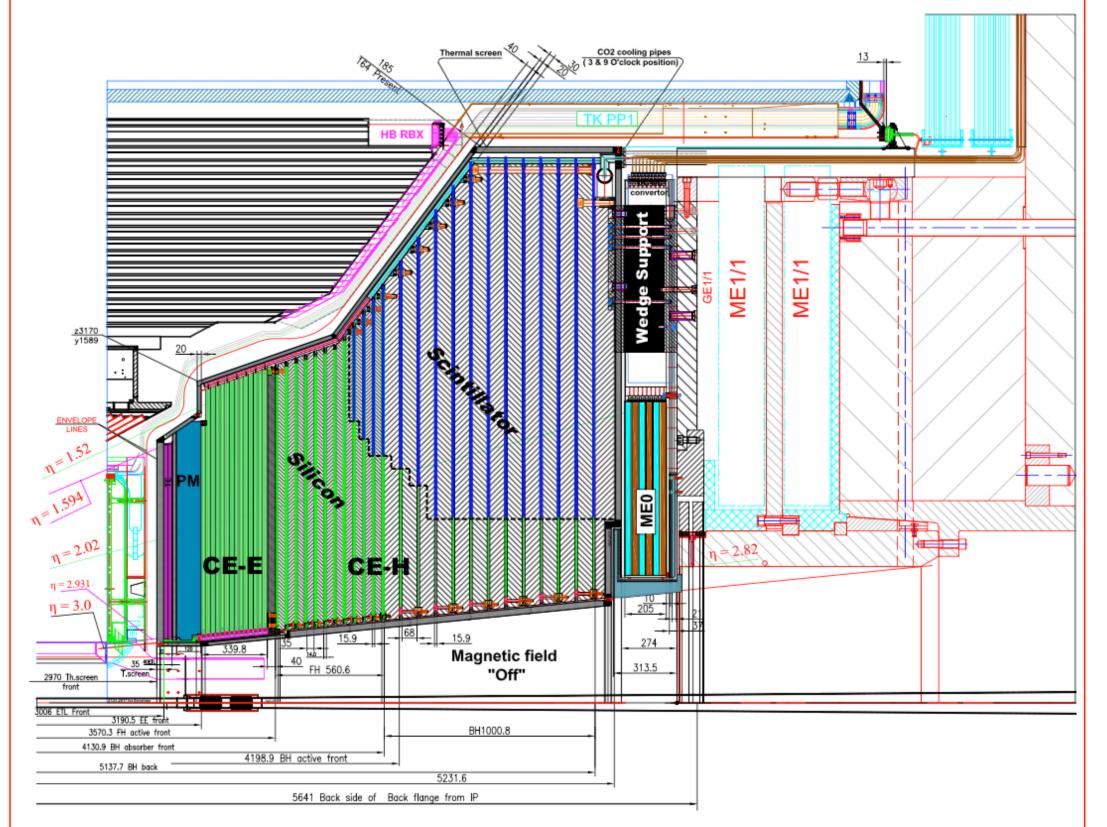
- Two categories:
  - Full size Copper cooling plate and  $CO_2$  (1)
  - Single Si module and Copper cooling plate (2)





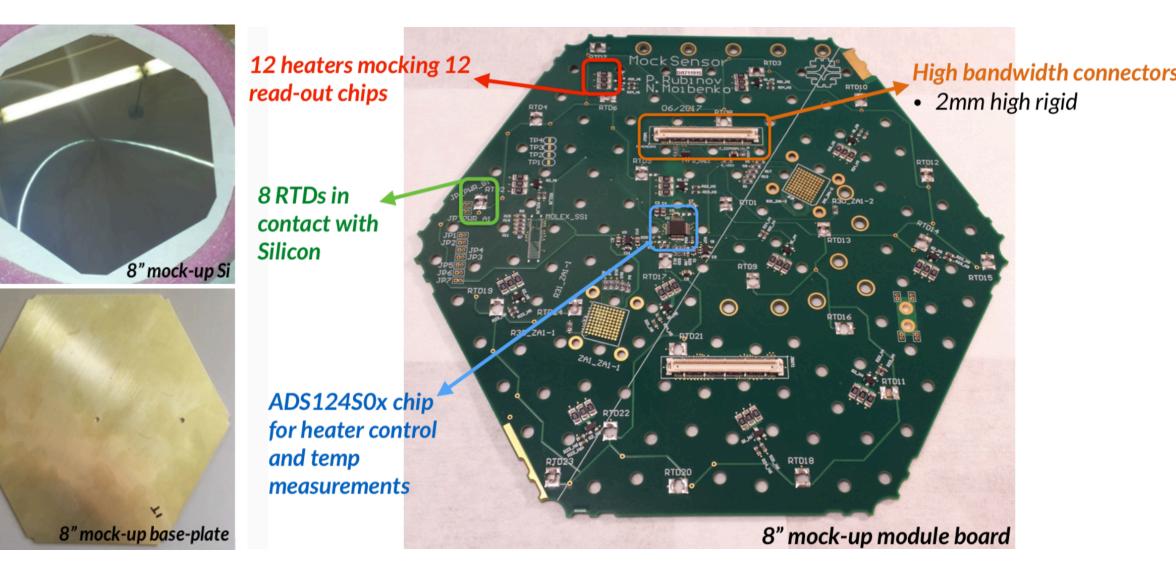


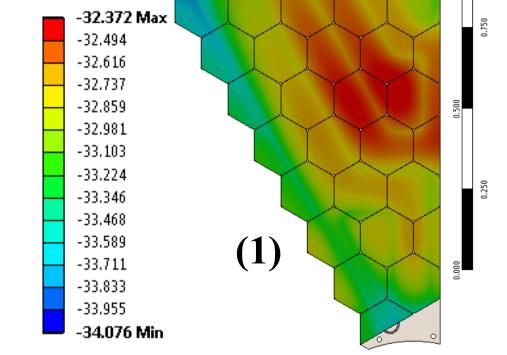
## **CMS High Granularity Calorimeter**



connection to the sensor back-plane through a conductive epoxy.

- Kapton itself provides electrical insulation to the sensor backplane from the baseplate.
- Mockup Silicon (Si) sensors:
- cut from 8" Si wafers & 750 um thick [design: 320 µm]
- Mockup module board (PCB):
  - 432 channel
  - Designed to apply heat loads & measure the Si temperature.
  - Enables performing thermal studies.
- \*\* Glue (Araldite 2011) layer is  $\sim 100 \mu m$  thick in each subsequent attachment.





1.7132 Max

1.6621

1.6111

1.56

1.5089

1.4579

1.4068

1.3558

1.3047

- 1.2536 Min

(2)

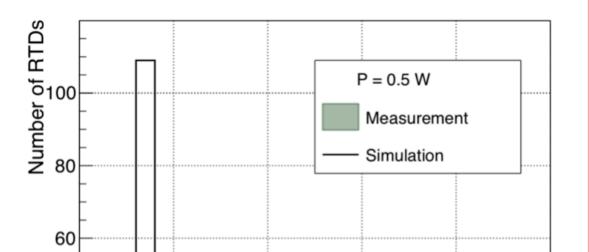
• 6.8 meter long tube

- 2.197 gm/sec CO<sub>2</sub> mass flow
- Pressure drop = 0.8 psi

At most the Copper plate is expected to be 2.5°C warmer than the  $CO_2$  (-35<sup>o</sup>C)

#### At Silicon level (Copper cooling plate and Silicon)

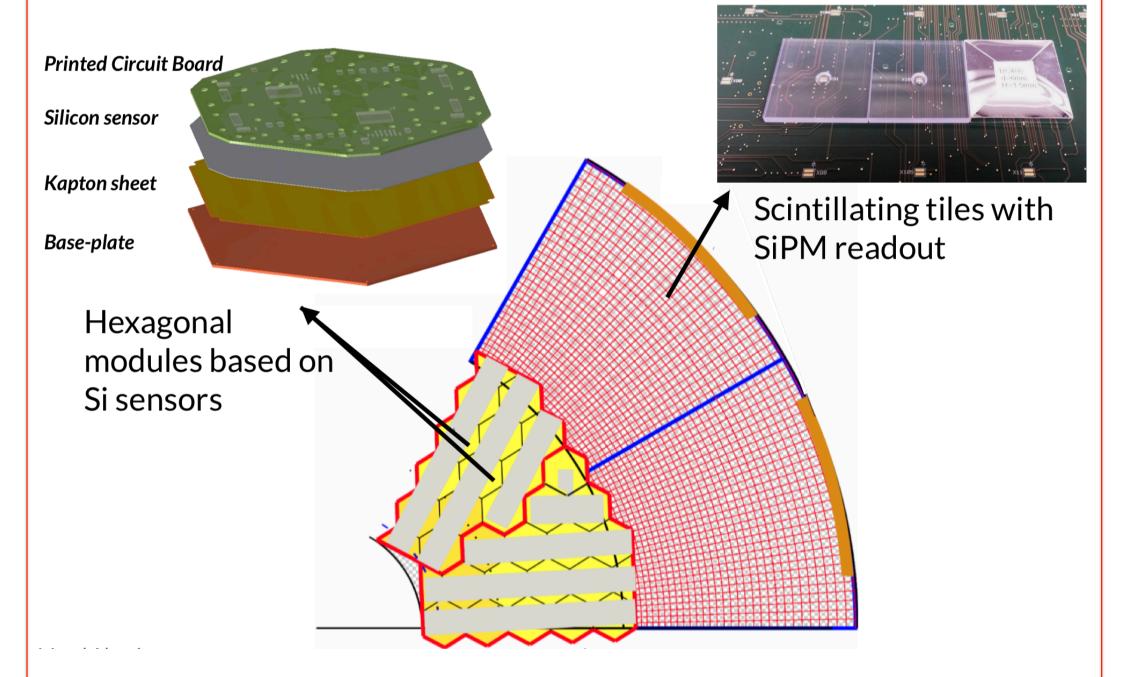
- 6 W of heat load is applied through 12 heaters (0.5 W each)
- On average the Si is expected to be 1.5°C warmer than the Copper cooling plate.
- Hence, all the Silicon sensors remain colder than  $-30^{\circ}C$ .



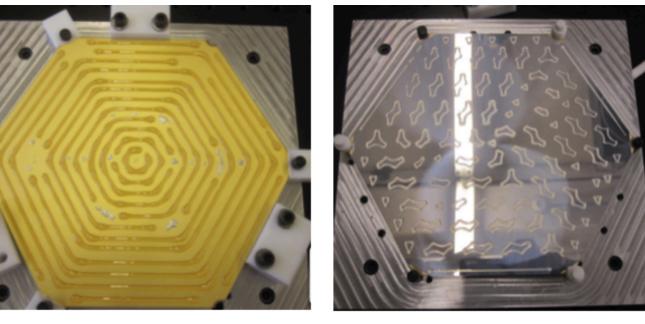
Measurement vs simulation

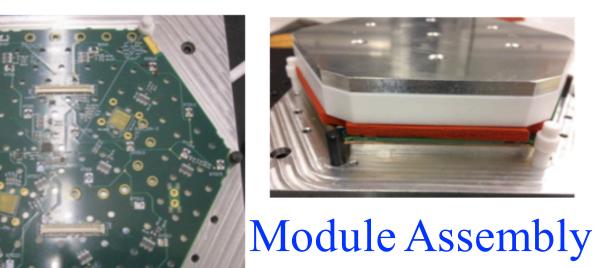
• 6 W of heat load is applied through 12 heaters (0.5 W each)

- A sampling calorimeter
- $\sim 500 \text{ m}^2 \text{ of scintillators}$
- $\sim 600 \text{ m}^2$  of silicon sensors
  - $\sim 6M$  channels
  - Expected to dissipate 220kW heat
- Operation ~  $-30^{\circ}$ C at sensor with the supply of CO<sub>2</sub> at  $-35^{\circ}$ C
- To reliably operate silicon sensors after irradiation, and  $\bullet$ to keep sufficiently low the energy equivalent of electronics noise from the increased leakage current & decreased charge collection efficiency after irradiation.
- Radiation damage is more nearer to the interaction point so, silicon sensor is used at the near side and plastic scintillator on the far side.



## **Mockup module & cassette assembly**







- 16 full modules
- 6 mother boards

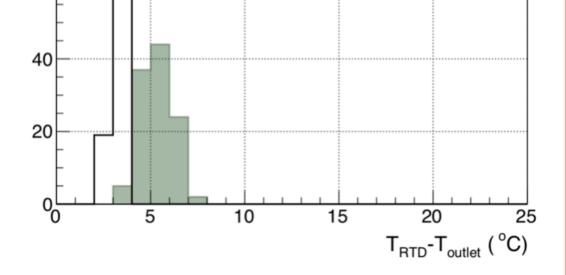
#### **Thermal test**

#### Temperature Readout:

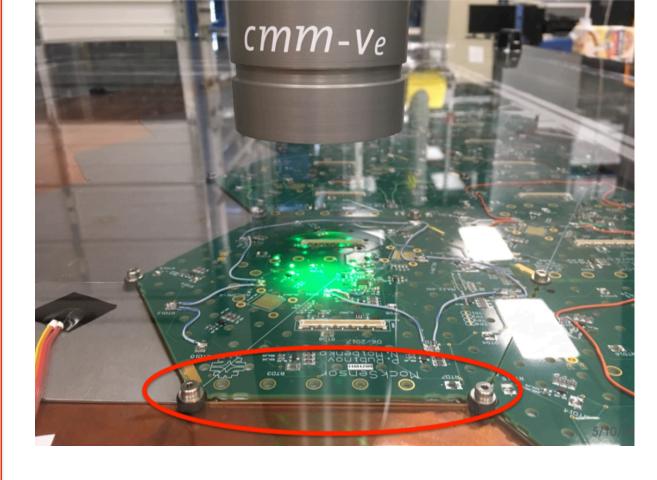
- We are reading 8 RTDs (4-wire connection) per module with ADS chip:
- 0.1°C temperature measurement precision; no need of calibration.

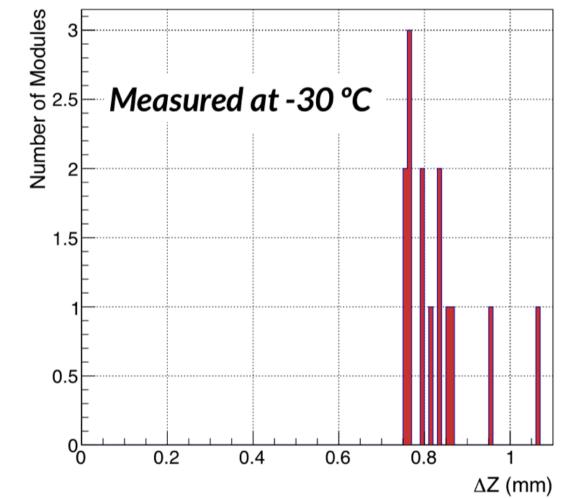
#### Initial gluing challenges in module construction:

• On average the measured temperature for Si was ~5.2°C above CO<sub>2</sub> outlet.



#### Module flatness at cold temperatures





- Coordinate Measurement Machine is used to measure the height of the center & 6 corners of the modules at cooling plate at  $-30^{\circ}$ C.
- Center of the modules is found to be  $\sim 800 \ \mu m$  higher than the corners.
- The air gap below the warped modules explains why the temperatures are larger than expected (simulation).
- Other base-plate material is being investigated that can prevent modules from warping.

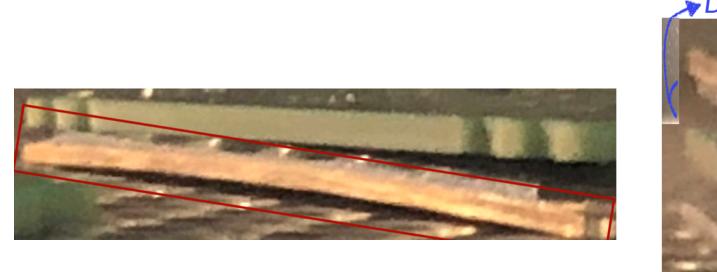
#### Conclusions

#### Purpose of the mockup test:

- To optimize the materials used in the construction of silicon-based modules for HGCAL.
- Thermal, mechanical prototype test.

#### CTE (10<sup>-6</sup>K<sup>-1</sup>) of brass baseplate, PCB, and Silicon: 12, 19, and 2.8.

- At -30°C, the brass baseplate and PCB are pulling the silicon sensor in  $\bullet$ opposite directions, & the glue has to resist the stress.
- 3 modules with insufficient amount of glue had the PCB delaminated & the sensor got cracked.
- After ensuring sufficient glue is applied, no more crack was observed.



- First milestone i.e. first thermo-mechanical prototype test (28-Aug-18) is very well within the reach.
- The first mockup modules & cassette for thermo-mechanical study are produced:
  - Various base-plate material is being investigated to safely accommodate the large differences in CTE compared to that of the Silicon sensors.
  - An extensive set of detailed simulations and tests with full scale realistic mock-ups is currently under way to further optimize this design and arrive at a fully engineered solution.

#### References

• CMS-TDR-019: The Phase-2 Upgrade of the CMS endcap calorimeter

#### Fermi National Accelerator Laboratory

